



DTI REPORT NO. 655

**HEAVYWEIGHT HIGH IMPACT (H.I.) SHOCK TEST REPORT
FOR THE ARGON SHIP EXPLORATION
EQUIPMENT INCREMENT E (SSEE-E) SYSTEM FOR 901D**

Prepared For:
901D, LLC
360 Route 59
Tallman, New York 10982-0615

Prepared Under:
Purchase Order No. S0250D

July, 2003

Publications Control No. R030704

Dynamic Testing • A Division of DTI Holdings, LLC • P.O. Box 494, 3915 Sunnymeade Rd., Rustburg, VA 24588 • 434-846-0244 • dti@dynatest.net

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1.0 INTRODUCTION

1.1 Purpose

This test report documents for 901D the requirements, responsibilities, and actions accomplished by Dynamic Testing (DTI) during the heavyweight, high-impact shock test of the Argon Ship Exploration Equipment Increment E (SSEE-E) System for 901D.

1.2 Background

The Argon Ship Exploration Equipment Increment E (SSEE-E) System is planned for initial installation on DDG, CG, and LHAS Class ships. The installation on LHD8 and LP Class ships requires Grade A shock qualification in accordance with MIL-S-901D.

1.3 Scope

This report describes the activities performed during the pre-test period, the fixture and machinery installation aboard the floating shock platform (FSP), actual test operations, instrumentation setup, physical inspections, and operational tests of Ship Exploration Equipment Increment E (SSEE-E) System.

1.4 Objective

The objective of this test series was to receive MIL-S-901D, Grade A shock qualification approval for Ship Exploration Equipment Increment E (SSEE-E) System.

2.0 APPLICABLE DOCUMENTS

2.1 Military Specification

MIL-S-901D (Navy)

"Military Specification, Shock Tests, H.I. (High Impact); Shipboard Machinery, Equipment and Systems, Requirements for," dated 17 March 1989

2.2 Commercial Documents

DTI Drawing No. J655

"Test Fixture for UNDEX testing against the SSEE-E Cabinets for 901D"

DTI Drawing No. J-910220

“Tunable Deck Fixture Platform for
Floating Shock Platform UNDEX
Testing”

Document No. D00231-QTP-001

“Shock and Vibration Test Procedure
Prepared For Argon Engineering’s
Ship Signal Exploitation Equipment
Increment E (SSEE-E),” dated 24
April 2003

3.0 TEST REQUIREMENTS

3.1 Shock Specification

MIL-S-901D (U.S. Navy)

3.2 Shock Grade

Grade A

3.3 Equipment Class

Class II

3.4 Mounting Location

Deck

3.5 Mounting Plane Aboard Ship

- 1) HF Receive – Back and Base
- 2) System Control – Back and Base
- 3) V/UHF Receive – Back and Base
- 4) CCOP – Back and Base

3.6 Mounting Orientation Aboard Ship

- 1) HF Receive – Unrestricted
- 2) System Control – Unrestricted
- 3) V/UHF Receive – Unrestricted
- 4) CCOP – Unrestricted

3.7 Number of Articles

4

3.8 Items Tested

1) HF Receive Cabinet Configuration

Manufacturer: 901D
Part No: 275032

Shock Isolators: Base – (6) 901D Part No. 70512-1
Sway – (2) 901D Part No. 70526-1

Qty.	Hardware Item	Vendor
1	AN/SSQ-137 (V) 1 Rack	901D
1	Patch Panel HF	ARGON
1	Distribution Unit, RF	ARGON
1	Printer Unit, Laser	ARGON
1	Tuner, HF	ARGON
1	Workstation, Intel (Jots3)	ARGON
1	Keyboard/Display Unit	ARGON
1	Digital Processing Unit	ARGON
1	Uninterruptible Power Supply	ARGON

2) System Control Cabinet Configuration

Manufacturer: 901D
Part No: 275033

Shock Isolators: Base – (6) 901D Part No. 70512-1
Sway – (2) 901D Part No. 70526-1

Qty.	Hardware Item	Vendor
1	AN/SSQ-137 (V) 1 Rack	901D
1	Time Unit, Precision	ARGON
1	Distribution Unit, Time Frequency	ARGON
1	Filter Unit, HF, V/UHF	ARGON
1	Interface Unit, INT/EXT	ARGON
1	Patch Panel Audio	ARGON
1	Receiver, R-2411	ARGON
1	Up Converter	ARGON
1	Workstation, Intel (Jots4)	ARGON
1	Keyboard, Display Unit	ARGON
1	Server, Unix (CUB)	ARGON
1	Server Unix (LH)	ARGON
1	Uninterruptible Power Supply	ARGON

3) V/UHF Receive Cabinet Configuration

Manufacturer: 901D
Part No: 275034

Shock Isolators: Base – (6) 901D Part No. 70512-1
Sway – (2) 901D Part No. 70526-1

Qty.	Hardware Item	Vendor
1	AN/SSQ-137 (V) 1 Rack	901D
1	Patch Panel, V/UHF	ARGON
1	Distribution Unit, RF, V/UHF	ARGON
1	Receiver, R-2412	ARGON
1	Receiver, V/UHF	ARGON
1	Workstation, Intel (Jots5)	ARGON
1	Keyboard/Display Unit	ARGON
1	Digital Processing Unit	ARGON
1	Uninterruptible Power Supply	ARGON

4) CCOP Cabinet Configuration

Manufacturer: 901D
Part No: 275035

Shock Isolators: Base – (6) 901D Part No. 70512-1
Sway – (2) 901D Part No. 70526-1

Qty.	Hardware Item	Vendor
1	AN/SSQ-137 (V) 1 Rack	901D
1	Uninterruptible Power Supply	ARGON

3.9 Overall Equipment Dimensions/Weight/Center of Gravity

1) HF Receive Cabinet

Dimensions: 71.97" H x 22.0" W x 34.62" D (with isolators)
Weight: 724 pounds
Vertical CG: 34" from mounting base

2) System Control Cabinet

Dimensions: 71.97" H x 22.0" W x 34.62" D (with isolators)
Weight: 820 pounds
Vertical CG: 34" from mounting base

3) V/UHF Receive Cabinet

Dimensions: 71.97" H x 22.0" W x 34.62" D (with isolators)
Weight: 824 pounds
Vertical CG: 34" from mounting base

4) CCOP Cabinet

Dimensions: 71.97" H x 22.0" W x 34.62" D (with isolators)
Weight: 297 pounds (dummy loaded to 800 lbs. for testing)
Vertical CG: 34" from mounting base

3.10 Mode of Operation

The test units were powered and operating for all detonations. The equipment under test (EUT) was powered with 110 VAC.

3.11 Shock Test Acceptance Criteria

In addition to the acceptance criteria presented in Section 3.1.10 of MIL-S-901D (Navy), the following was also considered as acceptance criteria for meeting the shock test requirements:

1. The test item continues to perform its intended functions following the test series. The proper operation of the test item was verified by successfully completing Argon's operational test procedure outlined in Appendix A of Document No. D00231-QTP-001," Shock and Vibration Test Procedure Prepared for Argon Engineering's Ship Signal Exploitation Equipment Increment E (SSEE-E)," dated 24 April 2003.
2. A momentary malfunction shall be considered acceptable if it is automatically self-correcting.
3. No part of the test item that may be considered a hazard to personnel or Grade A equipment shall break or come adrift.
4. The test item shall not demonstrate a potential for fire hazards. Any evidence of electrical shorts, release of flame, smoke, or sparks shall be cause for rejection unless otherwise approved by the acceptance authority.
5. Minor physical damage to the test item, such as small cracks, minor yielding of structure, out-of-tolerance clearances, and similar damage shall not be cause for shock test disapproval unless such damage causes unacceptable impairment of equipment performance, results in a hazard, or results in a substantially shortened equipment useful life.

3.12 FSP-Borne Weight

Test Items	3,168 pounds
Test Fixtures for Class II Items	3,573 pounds
Other Items on DSF	3,386 pounds
DSF	11,000 pounds
Instrumentation and Rack	1,200 pounds
Canopy	<u>7,000 pounds</u>

Total Weight Borne by FSP 29,327 pounds

4.0 TEST METHOD

4.1 Test Facility

The test series was conducted at the facilities of Dynamic Testing (DTI). This test facility is approved for testing in accordance with NAVSEAINST 9491.1B and by NAVSEA Letter 55X11/SH, Serial 87, dated 17 February 1984.

4.2 Test Series

The test series was conducted in accordance with MIL-S-901D and 901D Document No. D00231-QTP-001, using the FSP, constructed in accordance with BUSHIPS Drawing No 645-1973904, utilizing standard, 60-pound HBX charges suspended at a depth of 24 feet below the surface of the water. Standoff distances were measured from the center of the charge to the FSP vertical hull plating.

After Shot No. 3, the test items were rotated 90 degrees in order to meet requirements for unlimited orientation.

Table 1. Test Series

Shot/MIL-S-901D Test Number	Standoff Distance	Deck Frequency	Charge Location
1/2	30 feet	14 ± 2 Hz.	Along the athwartship centerline of the FSP
2/3	25 feet	14 ± 2 Hz.	Along the athwartship centerline of the FSP
3/4	20 feet	14 ± 2 Hz.	Along the athwartship centerline of the FSP
4/4 Rotated	20 feet	14 ± 2 Hz.	Along the athwartship centerline of the FSP

The fundamental vertical response of the DSF was determined from the peak response on a fast fourier transform (FFT) generated from acceleration time-history data measured at Sensor A1V (see Photograph No.3). Deck frequency and support locations are summarized in Table 2.

Table 2. DSF Frequency and Pin Support Locations

Shot/Test No.	Sensor No.	Frequency (Hz.)	DSF Pin Support Locations
1/2	A1V	13.1	A5/D5
2/3	A1V	13.1	A5/D5
3/4	A1V	13.1	A5/D5
4/4	A1V	13.1	A5/D5

4.3 Exceptions to MIL-S-901D

Instead of the normal sequence of shots specified in MIL-S-901D, the shot sequence listed in Table 3 was conducted in order to meet requirements for unrestricted orientation.

Table 3. Planned Shot Sequence

Shot No.	MIL-S-901D Test No.	Charge Standoff (feet)
1	2	30
2	3	25
3	4	20
4	4 (rotated)	20

5.0 TEST INSTALLATION/CONFIGURATION

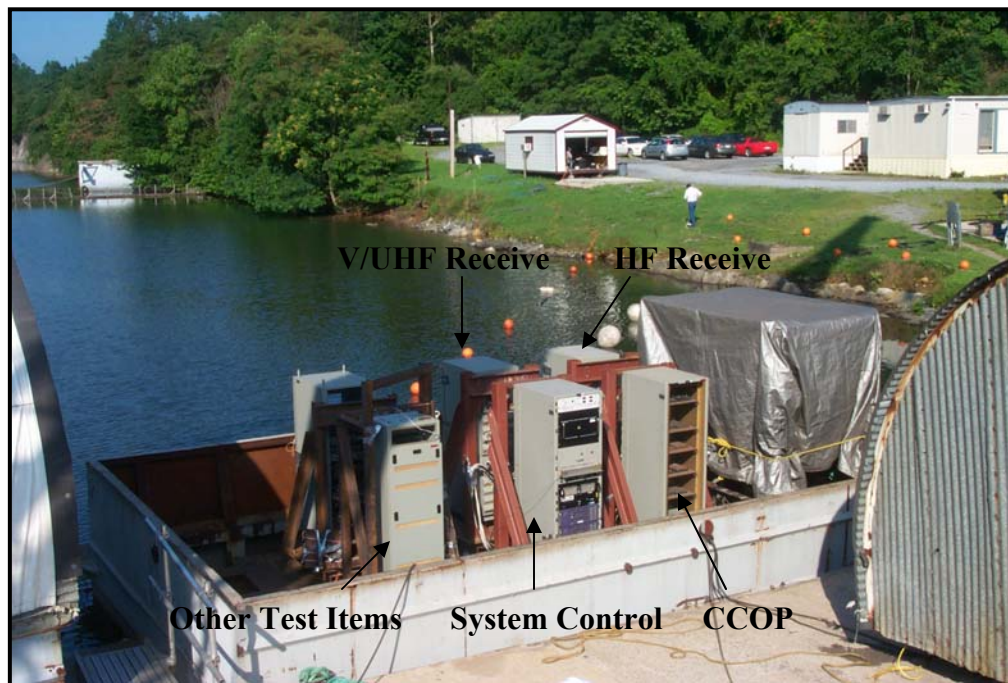
5.1 Equipment Installation

Photograph No. 1 shows the installation for Shots 1 through 3.

Photograph No. 2 shows the installation for Shot 4.



Photograph No. 1. Overhead View of Initial Installation, Shots 1-3



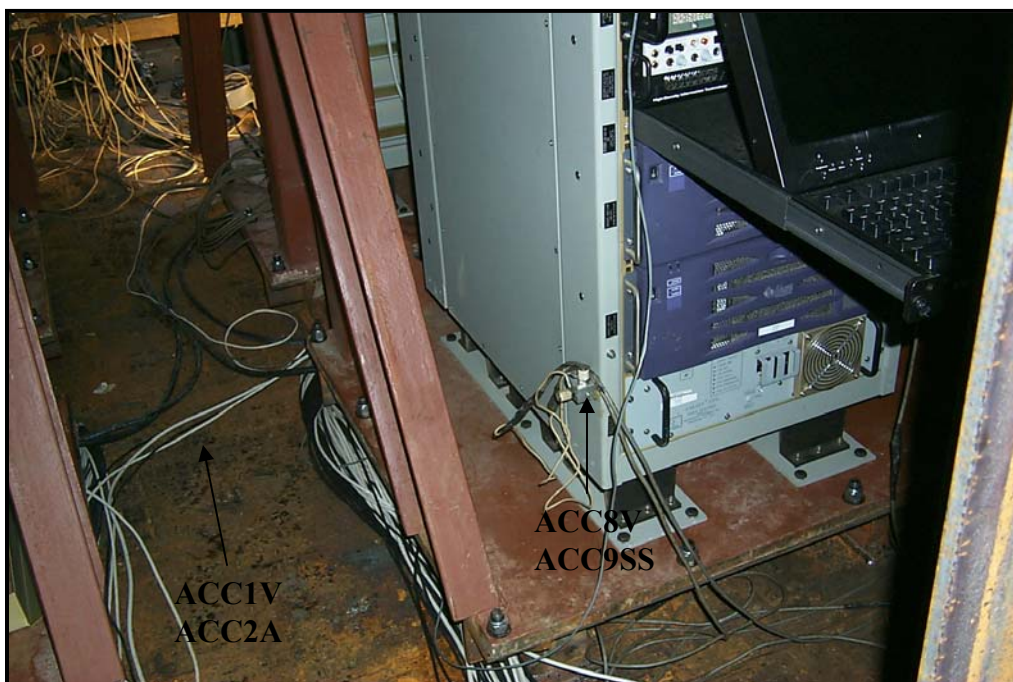
Photograph No. 2. Overhead View of Initial Installation, Shot 4

5.2 Test Instrumentation

One velocity meter (VM), and nine accelerometers (ACC) were installed on the equipment and FSP during the test series to verify proper test geometries. One digital recording in Windows Media Player format was taken of each shot. Location of instrumentation listed in Table 4 can be seen in Photograph Nos. 3 through 7.

Table 4. Test Instrumentation

Item	Orientation	Location
VM1	Vertical	FSP Innerbottom, Blast Side
A1V	Vertical	Center DSF
A2A	Athwartship	Center DSF
A5V	Vertical	Top Right Front of HF Receive Rack
A6V	Vertical	Top Right Front of System Control Rack
A7SS	Side – Side	Top Right Front of System Control Rack
A8V	Vertical	Bottom Left Front of System Control Rack
A9SS	Side - Side	Bottom Left Front of System Control Rack
A10V	Vertical	Top Right Front of V/UHF Receive Rack
A11V	Vertical	Top Right Front of CCOP Rack



Photograph No. 3. Instrumentation Locations



Photograph No. 4. Instrumentation Location



Photograph No. 5. Instrumentation Location



Photograph No 6. Instrumentation Location



Photograph No. 7. Instrumentation Location

6.0 TEST RESULTS

6.1 Test Schedule

Table 5. Test Schedule Data

Shot No./Test No.	Standoff	Date	Time (EST)
1/2	30' Athwartship	7-July-03	1222
2/3	25' Athwartship	8-July-03	1159
3/4	20' Athwartship	10-July-03	1231
4/4	20' Athwartship	11-July-03	1546

6.2 Survey Findings

Pre- and post-test inspections and operational verifications were conducted before and after each shock test in accordance with 901D Document No. D00231-QTP-001, dated 24-April-2003.

6.2.1 Survey Findings Prior to Testing

6.2.1.1 **Action:** The equipment under test (EUT) was inspected for proper installation and visual deficiencies.

Observation 1: No issues.

Resolution 1: N/A

6.2.1.2 **Action:** The EUT had operational tests performed by the onsite Argon representative.

Observation 1: No issues.

Resolution 1: N/A

6.2.2 Survey Findings after Shot 1, MIL-S-901D, 30-foot Standoff

6.2.2.1 **Action:** The EUT was inspected for proper installation and visual deficiencies.

Observation 1: No issues.

Resolution 1: N/A

6.2.2.2 **Action:** The EUT had operational tests performed by the onsite Argon representative.

Observation 1: The KVM and keyboard were not operational in the System Control Rack.

Resolution 1: The front cover was removed and power cycled to the KVM and keyboard. This corrected the KVM and keyboard issues.

Observation 2: The HF Receive, V/UHF Receive, and System Control Racks were not operational after the shock event.

Resolution 2: The reset button on the System Control Rack was pushed to cycle power. This did not correct any of the system failures. The AC power was cycled on both servers and full operation was restored in approximately 9 minutes 45 seconds.

Observation 3: No further issues.

Resolution 3: N/A

6.2.3 Survey Findings after Shot 2, MIL-S-901D, 25-foot Standoff

6.2.3.1 **Action:** The EUT was inspected for proper installation and visual deficiencies.

Observation 1: No issues.

Resolution 1: N/A

6.2.3.2 **Action:** The EUT had operational tests performed by the onsite Argon representative.

Observation 1: The KVM and keyboard was not operational in the System Control Rack.

Resolution 1: The front cover was removed and power cycled to the KVM and keyboard. This corrected the KVM and keyboard issues.

Observation 2: The HF Receive, V/UHF Receive, and System Control Racks were not operational after the shock event.

Resolution 2: The reset button on the System Control Rack was pushed to cycle power. This did not correct any of the system failures. The AC power was cycled on both servers and full operation was restored in approximately 9 minutes 45 seconds.

Observation 3: No further issues.

Resolution 3: N/A

NOTE 1: The UPSs on only two of the four cabinets were operational during the first two shots. The on-site Argon representatives installed new ones, as seen in Photograph Nos. 8 and 9.

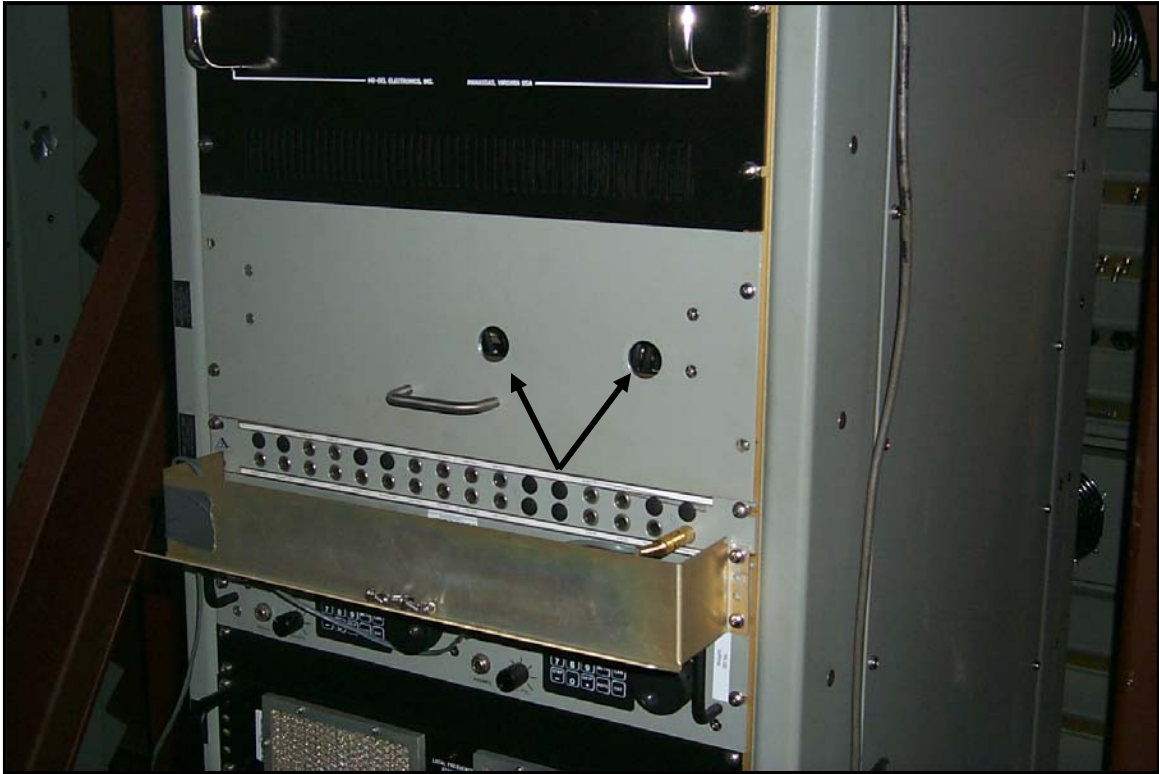


Photograph No. 8. New UPS



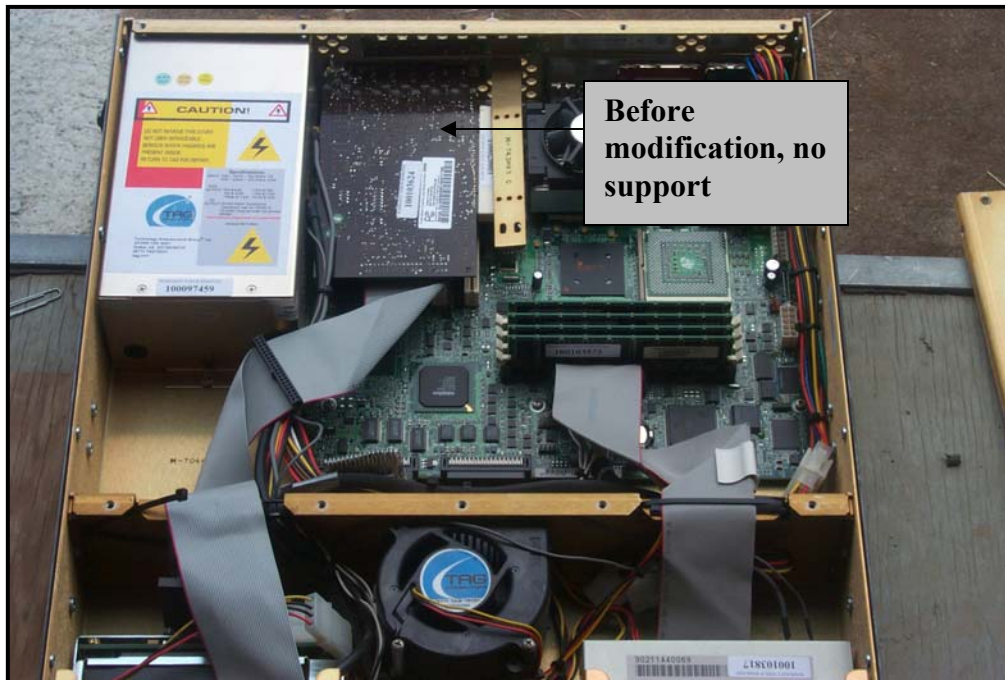
Photograph No. 9. Location of New UPS in Racks

NOTE 2: To enable the operator of the rack to reset the KVM, the cover was removed and modified, as seen in Photograph No. 10. Two 1-inch holes were drilled into the cover to allow for operator reset.



Photograph No. 10. Modification to KVM Cover

NOTE 1: The on-site Argon representatives installed a retaining bracket for the Audigy Sound Card to prevent deflection and/or movement. Photograph No. 11 shows the system before the bracket was added and Photograph No. 12 shows the system after the modification.



Photograph No. 11. Before Modification



Photograph No. 12. After Modification

6.2.4 Survey Findings after Shot 3, MIL-S-901D, 20-foot Standoff,

- 6.2.4.1 Action:** The EUT was inspected for proper installation and visual deficiencies.

Observation 1: No issues

Resolution 1: N/A

6.2.4.2 Action: The EUT had operational tests performed by the onsite Argon representative.

Observation 1: The R-2412 Circuit Breaker in the V/UHF Receive Rack had tripped.

Resolution 1: The circuit breaker was reset.

Observation 2: The V/UHF Receive and System Control Racks were not operational after the shock event.

Resolution 2: The racks were not operational because the UPS in the V/UHF Rack was off-line. The UPS was reset but did not operate. The UPS was by-passed and all EUT were operational.

Observation 3: No further issues.

Resolution 3: N/A

NOTE: The EUT was rotated 90° after completion of the post-shot inspection. Photograph No. 4 shows the rotated test setup.

6.2.5 Survey Findings after Shot 4, MIL-S-901D, 20-foot Standoff,

6.2.5.1 Action: The EUT was inspected for proper installation and visual deficiencies.

Observation 1: No issues.

Resolution 1: N/A

6.2.5.2 Action: The EUT had operational tests performed by the onsite Argon representative.

Observation 1: The HF Receive Rack was not operating correctly.

Resolution 1: Rack was rebooted and operated correctly.

Observation 2: The V/UHF Receive Rack was not powered properly. UPS was not working.

Resolution 2: The on-site Argon representative advised that a complete post-teardown inspection was required to troubleshoot the problems.

Observation 3: The System Control Rack Servers were not operating correctly.

Resolution 3: Servers were re-booted and operated correctly.

Observation 4: No further issues.

Resolution 4: This completed the test series.

7.0 POST-TEST ACTIONS/REQUIREMENTS

Upon completion of the shock test series, the equipment was returned to the following location where a breakdown inspection of each unit will be conducted in accordance with Paragraph 3.1.9.2 of MIL-S-901D:

Argon Engineering
Fairfax, Virginia

8.0 PERSONNEL PRESENT

Inspector	Representing	Shot No(s).
Tim Nogosky	DTI	1, 2, 3, 4
Mike Pearson	DTI	1, 2, 3, 4
Calvin Milam	DTI	1, 2, 3, 4
Steven Thompson	DTI	1, 2, 3, 4
John Aldon	901D	1, 2, 3
Sean Clubb	Argon	1, 2, 3, 4
Sandra Ebron	Argon	1, 2, 3, 4
Dan Umphrey	Argon	1, 2, 3, 4

9.0 SIGNATURES

Prepared by: Steven T. Thompson
Steven T. Thompson

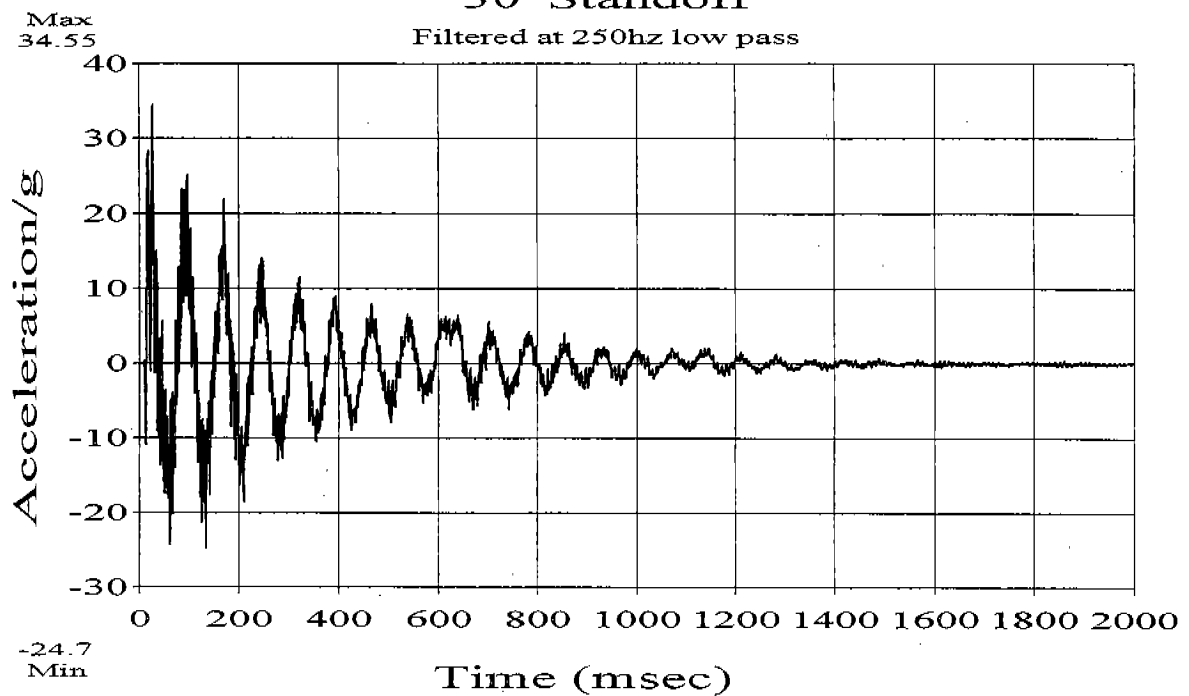
Reviewed by: Calvin P. Milam
Calvin P. Milam, Chief Engineer

Approved by: R.D. Fairfield
R.D. Fairfield, Vice President & General Manager

Appendix A
Test Instrumentation Data

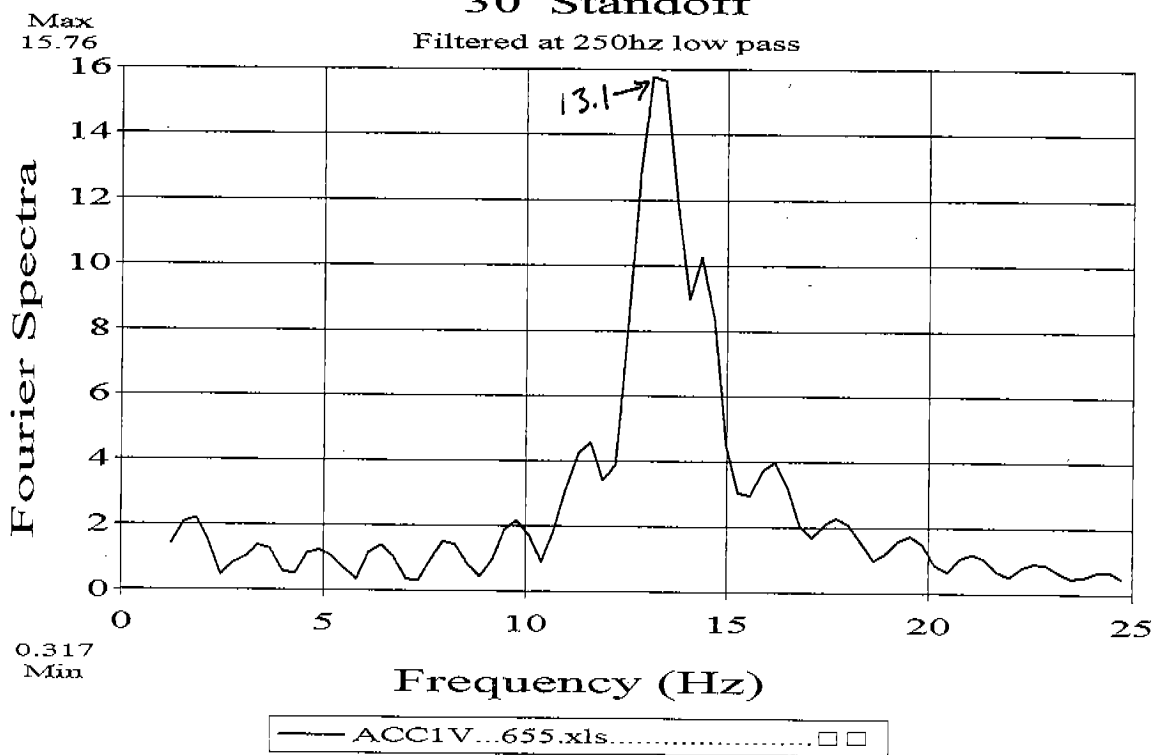
Shot1_Test2_J655

30' Standoff



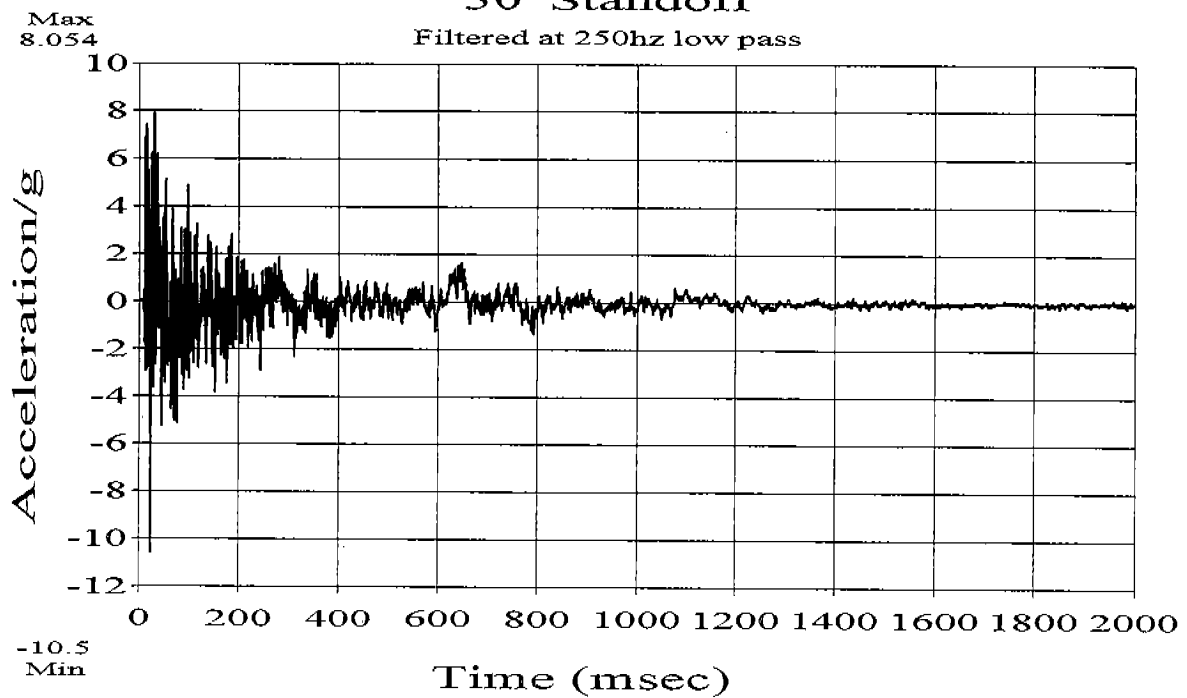
Shot1_Test2_J655

30' Standoff



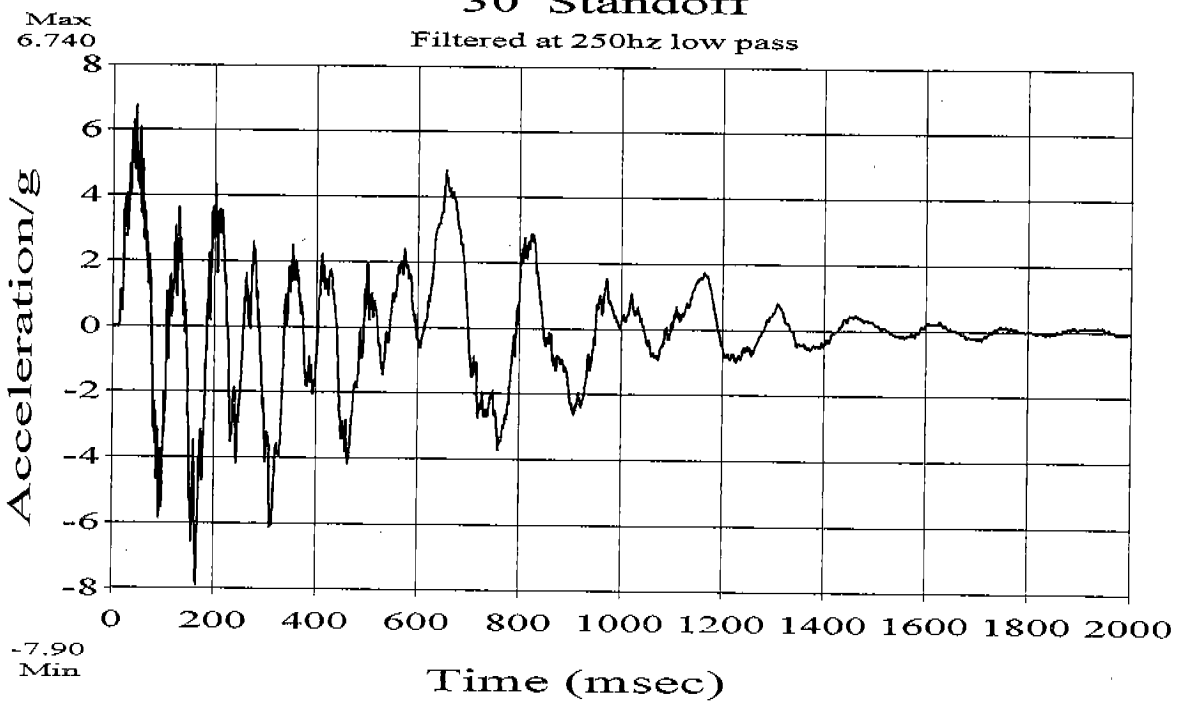
Shot1_Test2_J655

30' Standoff



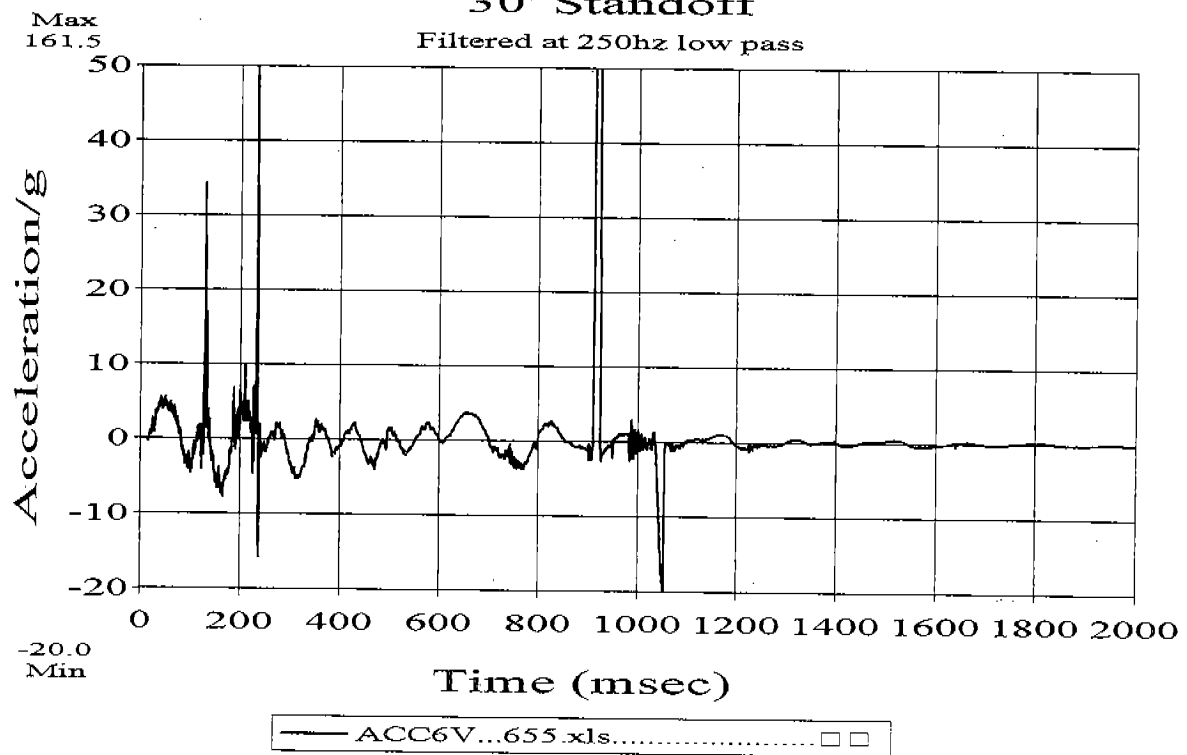
Shot1_Test2_J655

30' Standoff



Shot1_Test2_J655

30' Standoff

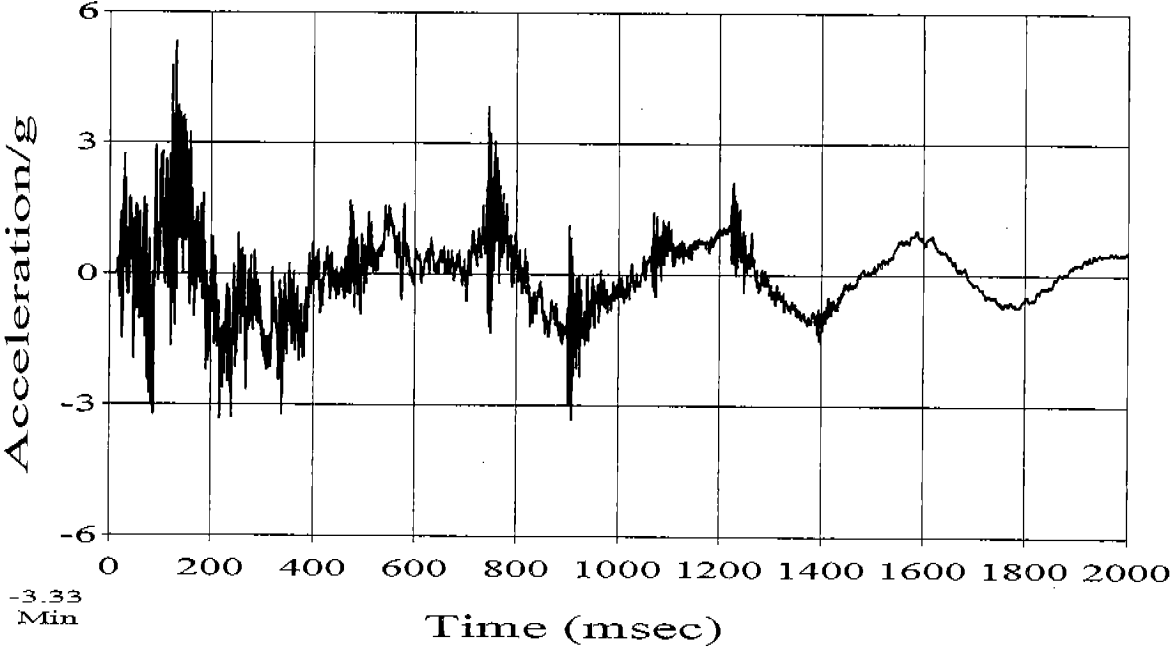


Shot1_Test2_J655

30' Standoff

Max
5.318

Filtered at 250hz low pass

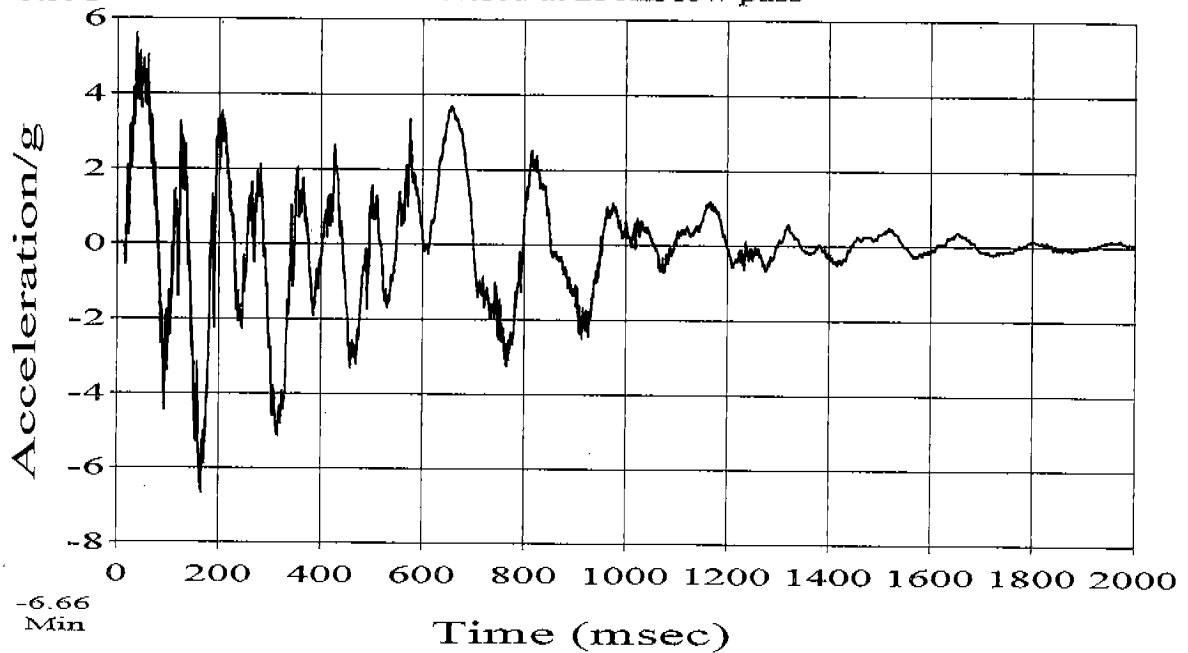


Shot1_Test2_J655

30' Standoff

Max
5.591

Filtered at 250hz low pass



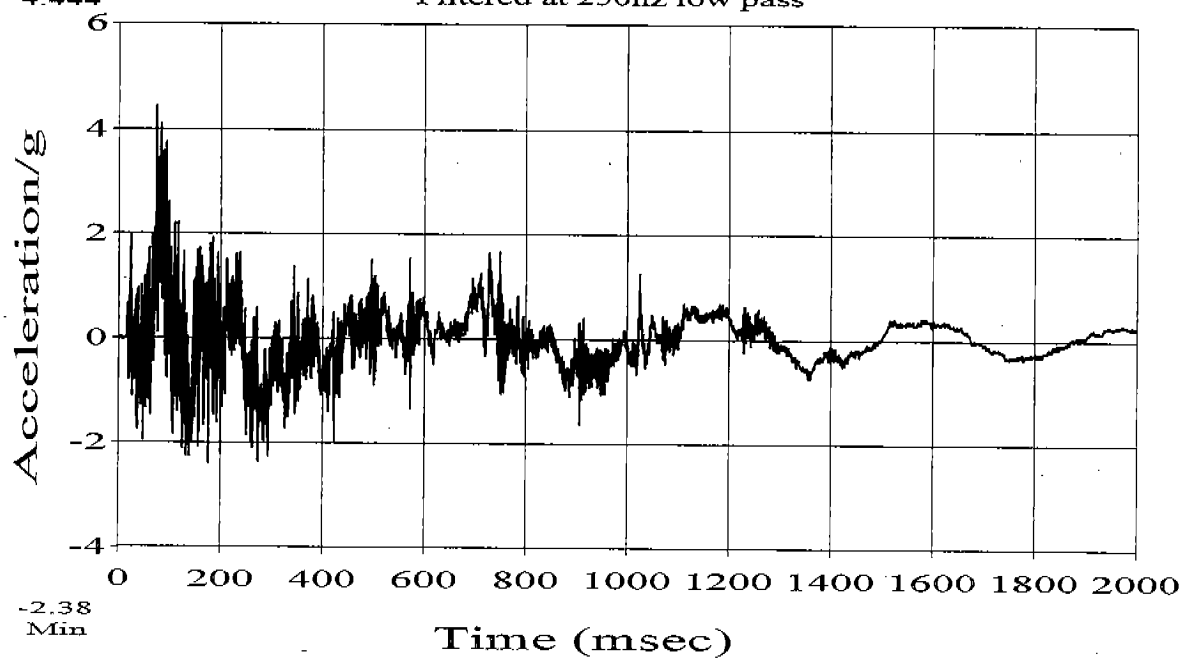
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Shot1_Test2_J655

30' Standoff

Max
4.444

Filtered at 250hz low pass



-2.38
Min

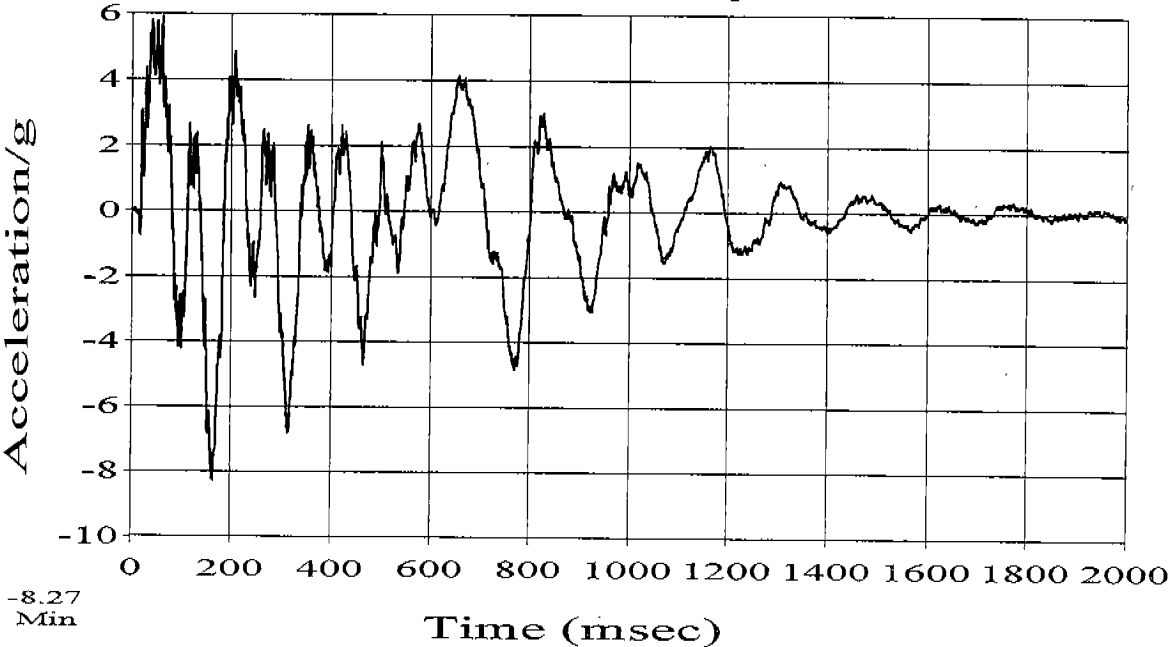
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Shot1_Test2_J655

30' Standoff

Max
5.904

Filtered at 250hz low pass

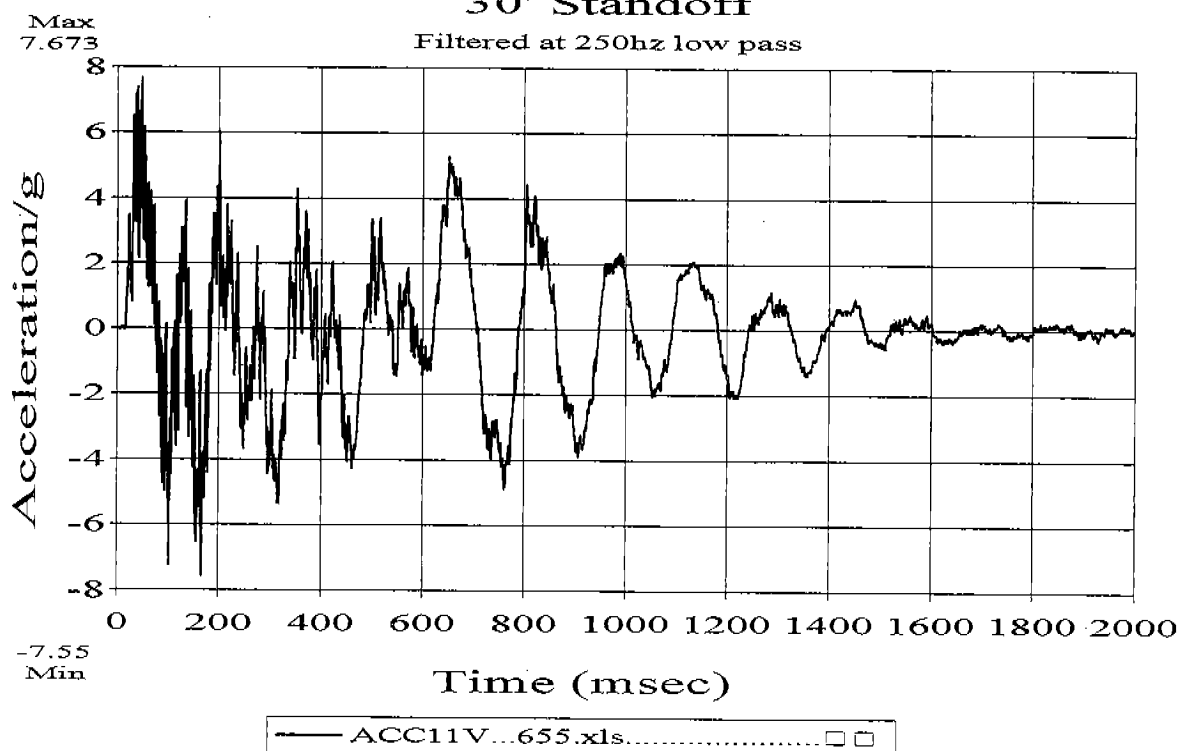


-8.27
Min

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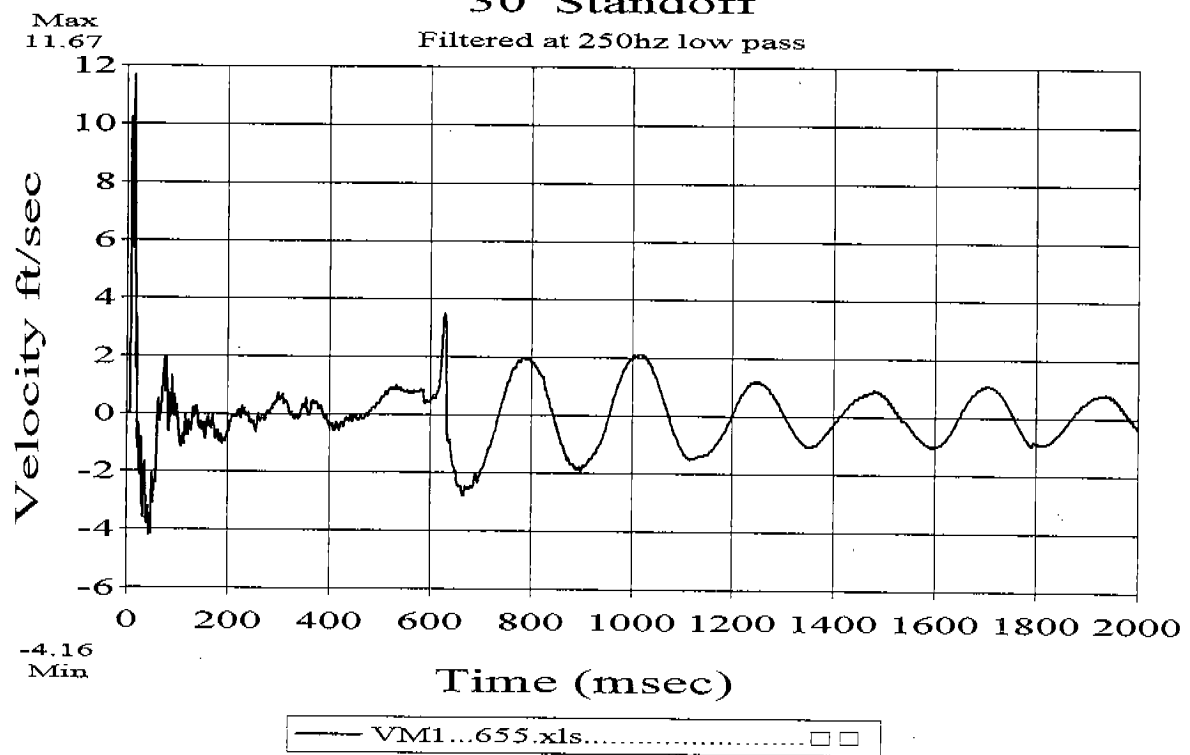
Shot1_Test2_J655

30' Standoff



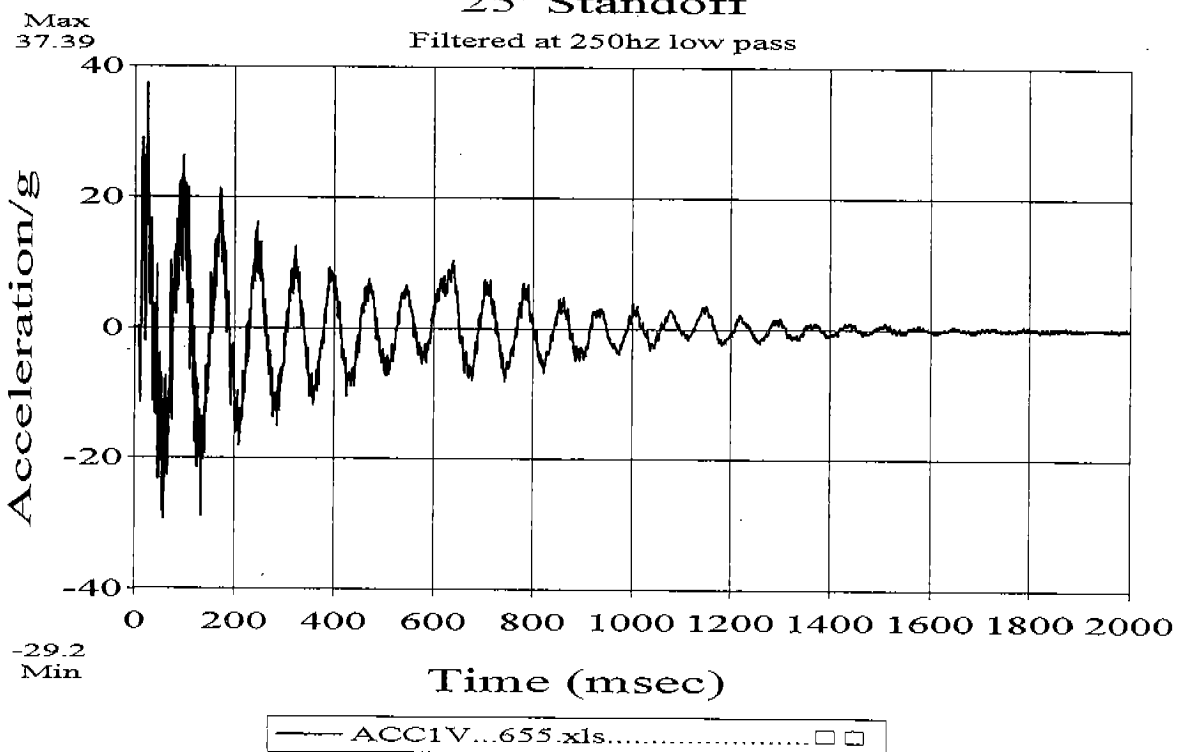
Shot1_Test2_J655

30' Standoff



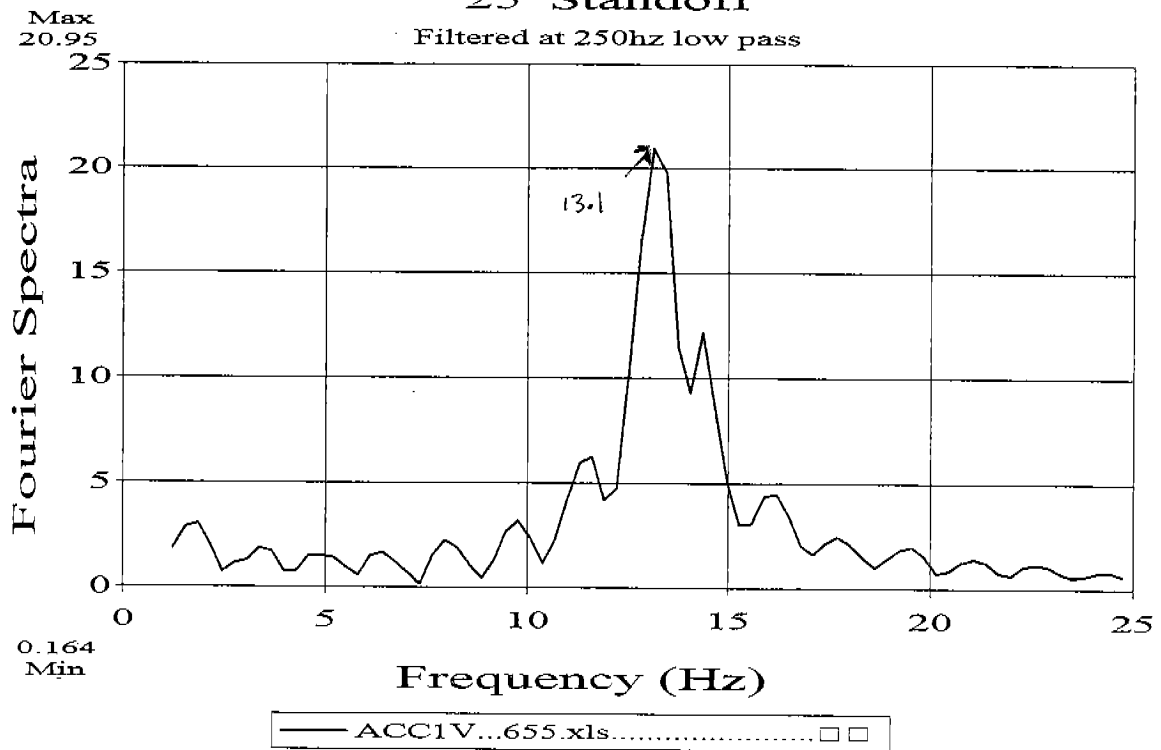
Shot2_Test3_J655

25' Standoff



Shot2_Test3_J655

25' Standoff

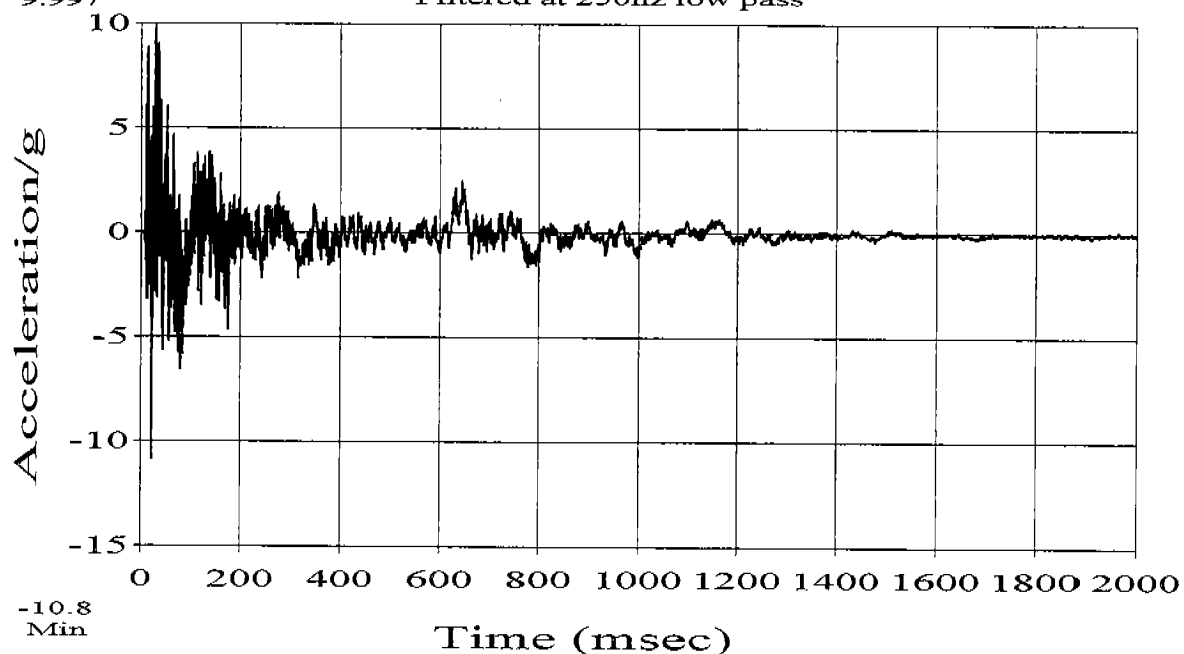


Shot2_Test3_J655

25' Standoff

Max
9.997

Filtered at 250hz low pass



-10.8
Min

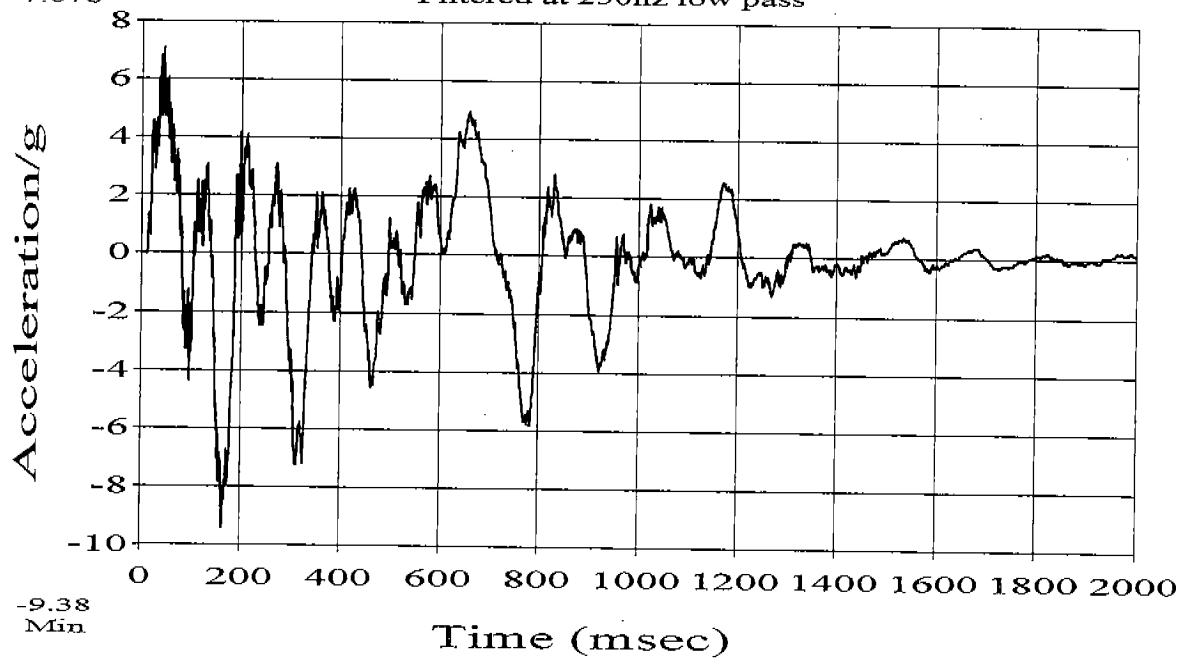
ACC2A...655.xls.....

Shot2_Test3_J655

25' Standoff

Max
7.078

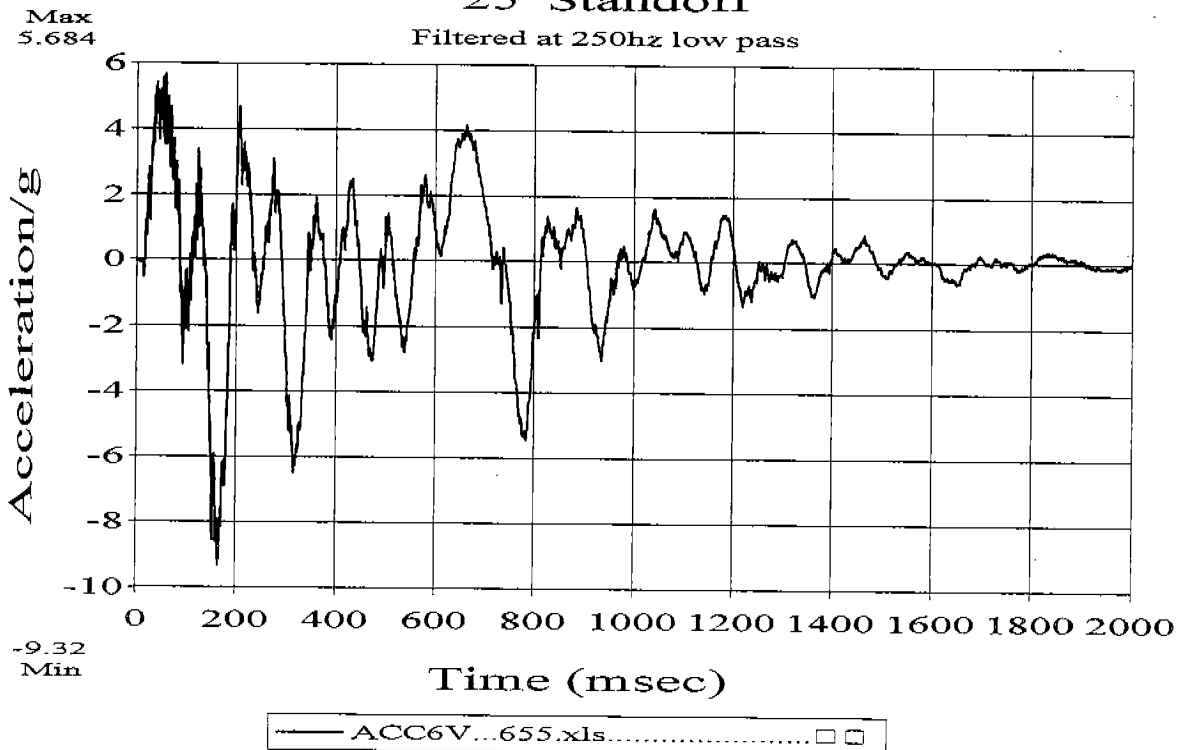
Filtered at 250hz low pass



— ACC5V...655.xls..... □ □

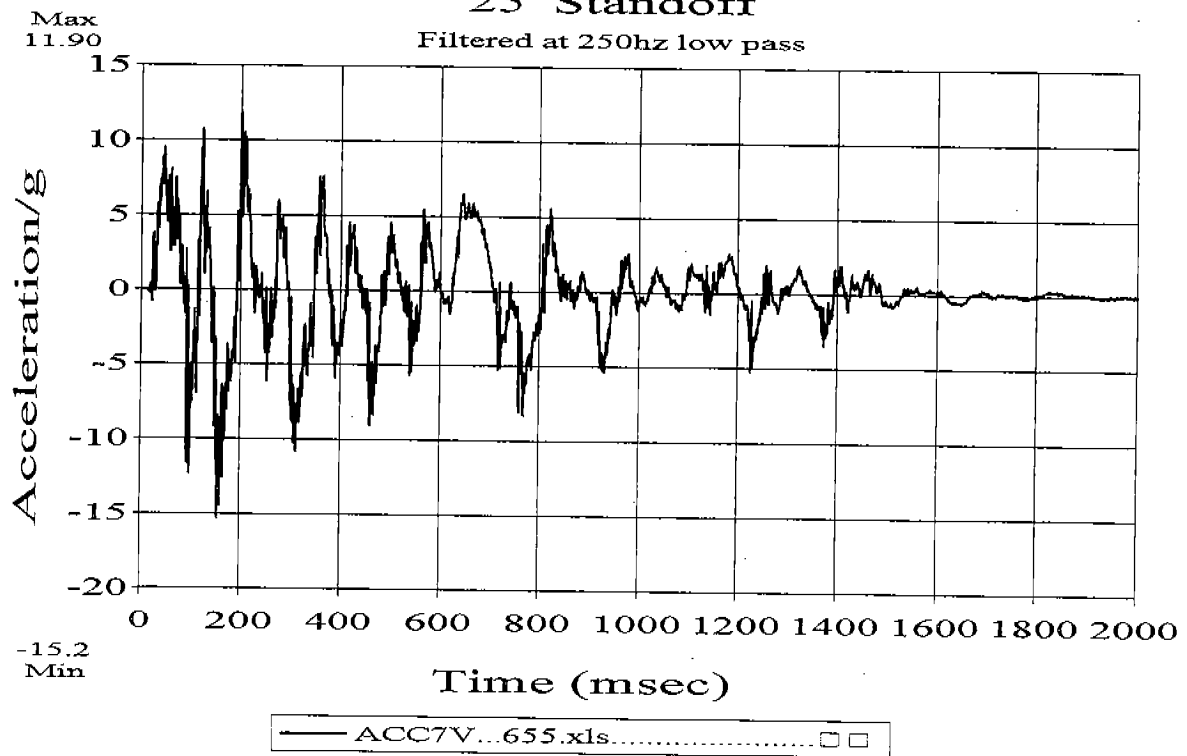
Shot2_Test3_J655

25' Standoff



Shot2_Test3_J655

25' Standoff

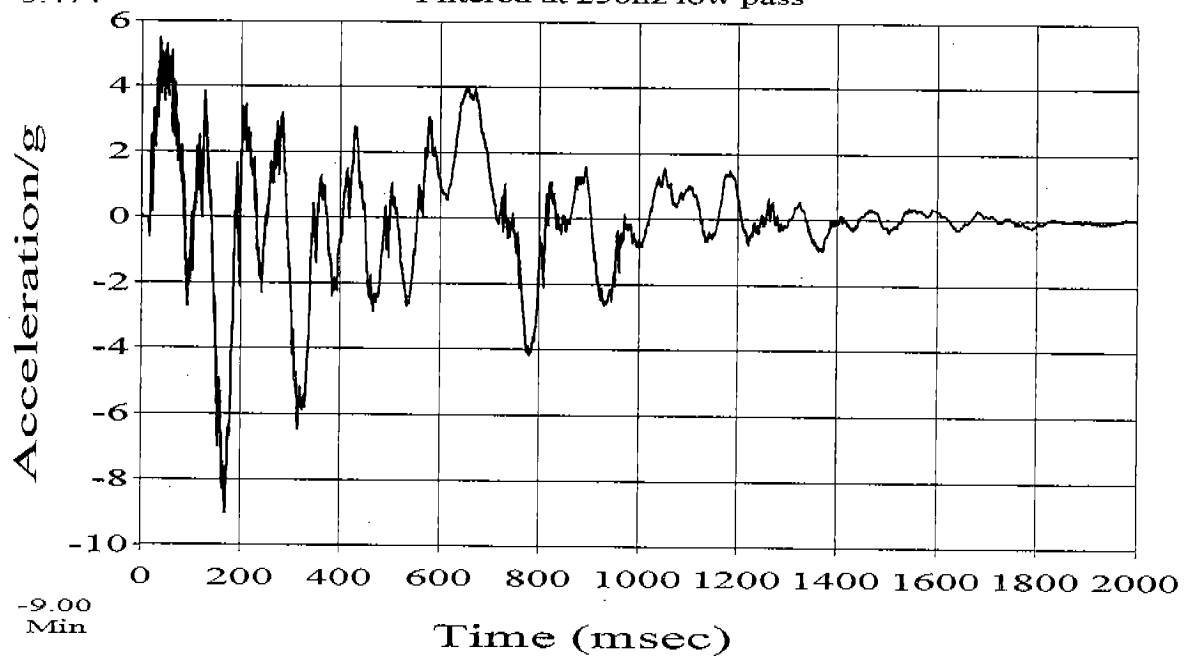


Shot2_Test3_J655

25' Standoff

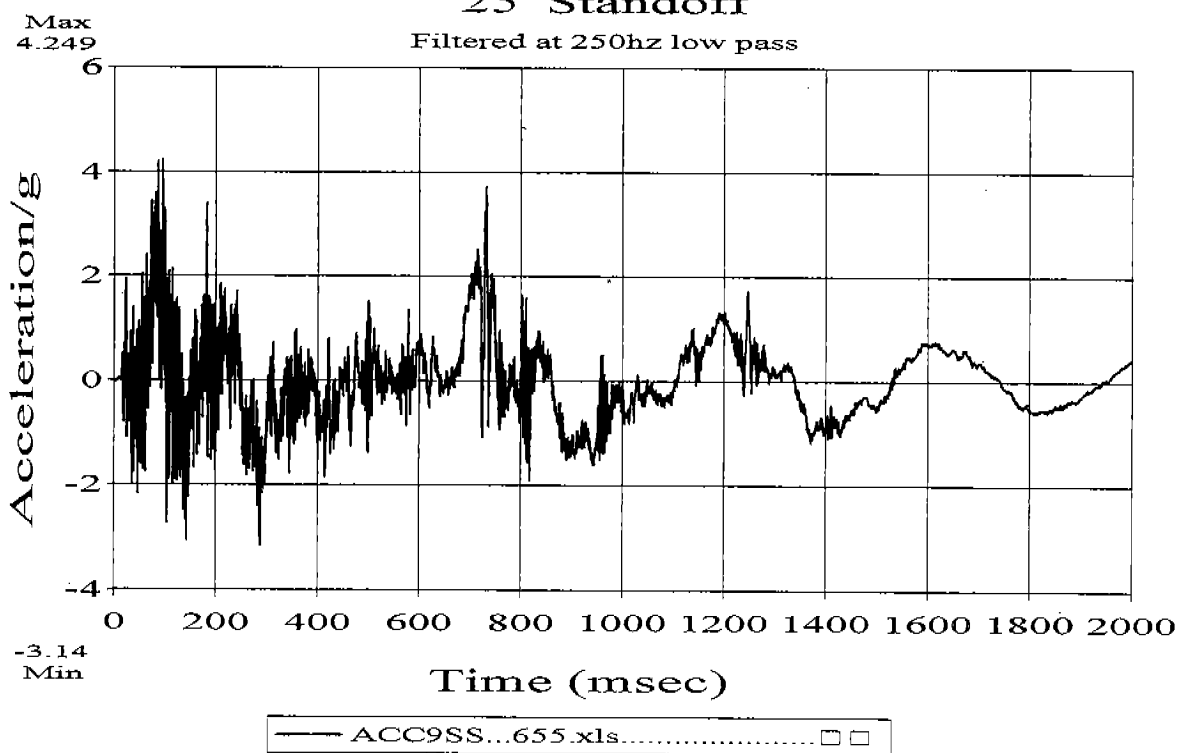
Max
5.474

Filtered at 250hz low pass



Shot2_Test3_J655

25' Standoff

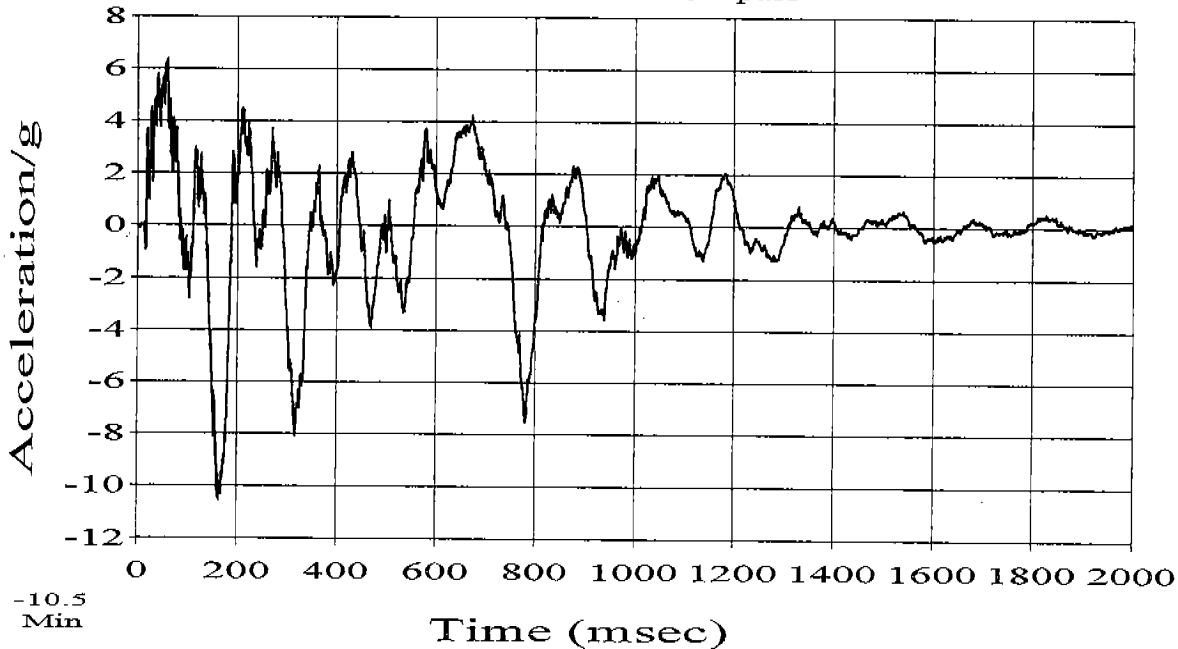


Shot2_Test3_J655

25' Standoff

Max
6.383

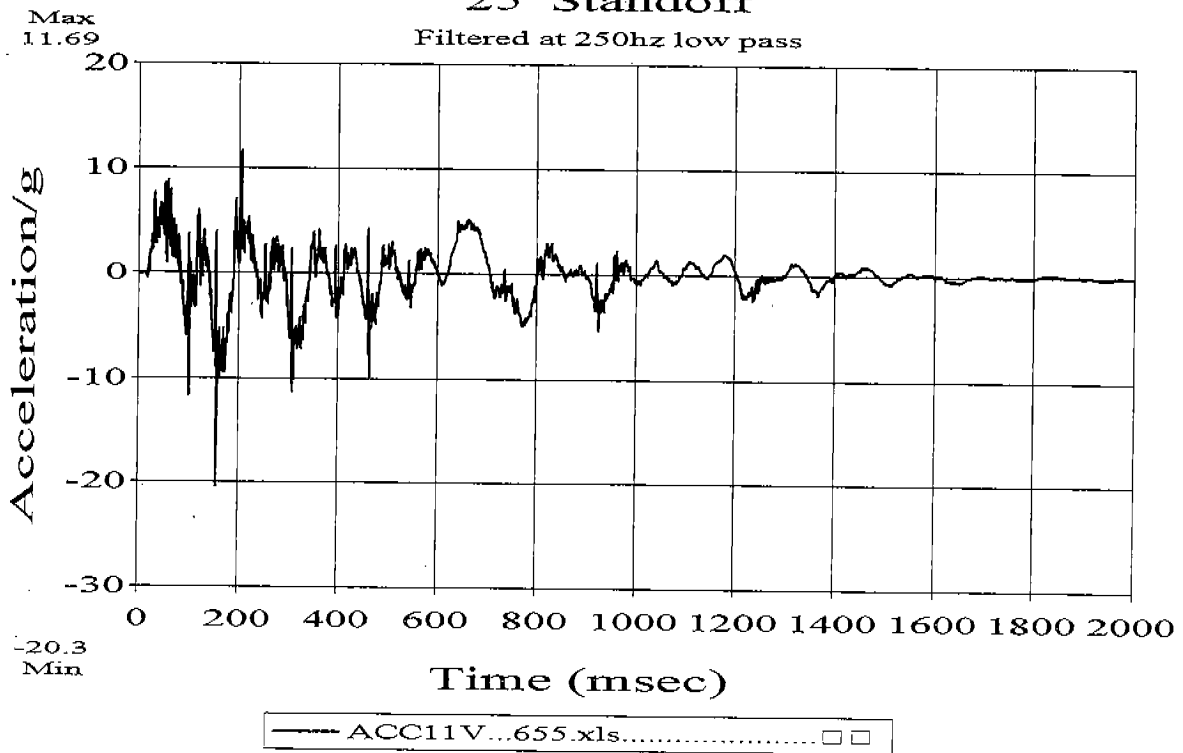
Filtered at 250hz low pass



ACC10V...655.xls.....

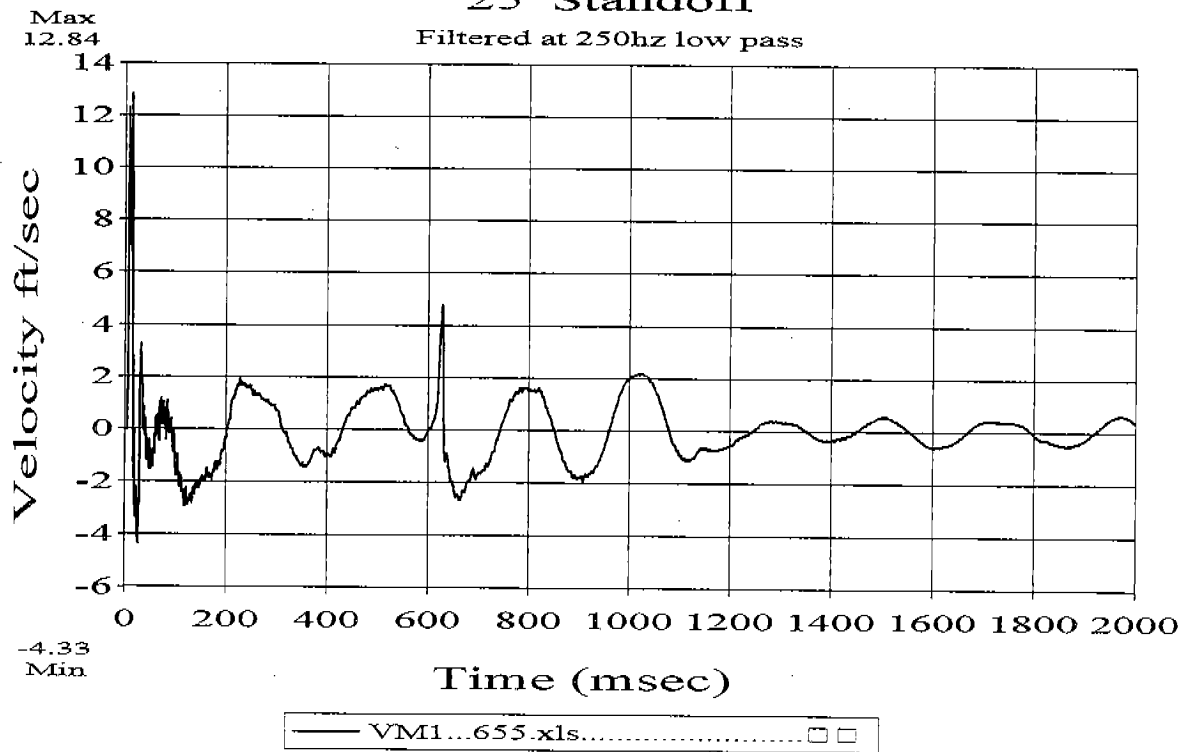
Shot2_Test3_J655

25' Standoff



Shot2_Test3_J655

25' Standoff

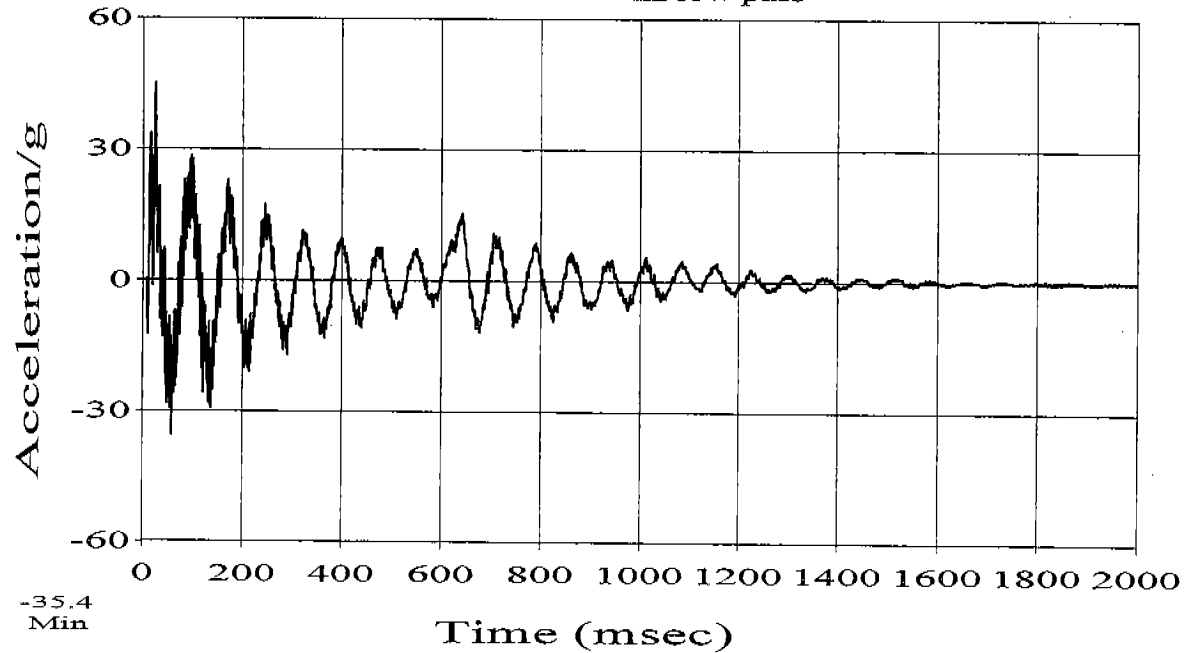


Shot3_Test4_J655

20' Standoff

Max
45.13

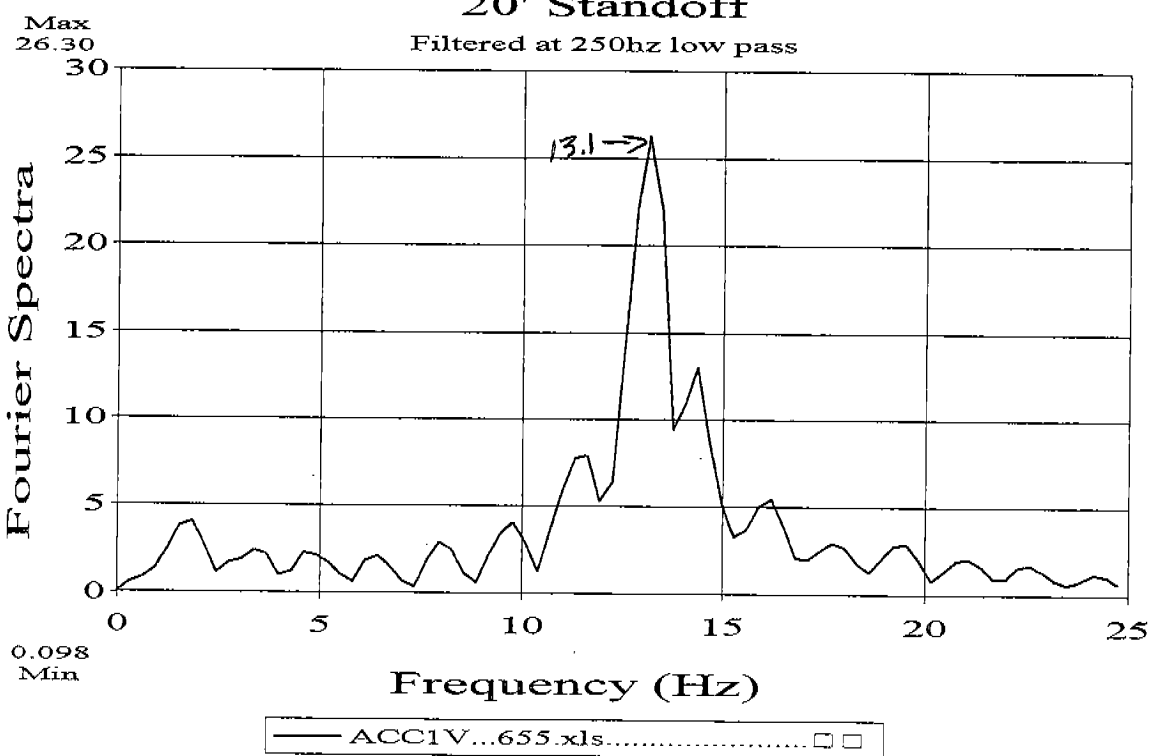
Filtered at 250hz low pass



— ACC1V...655.xls..... ☐ ☐

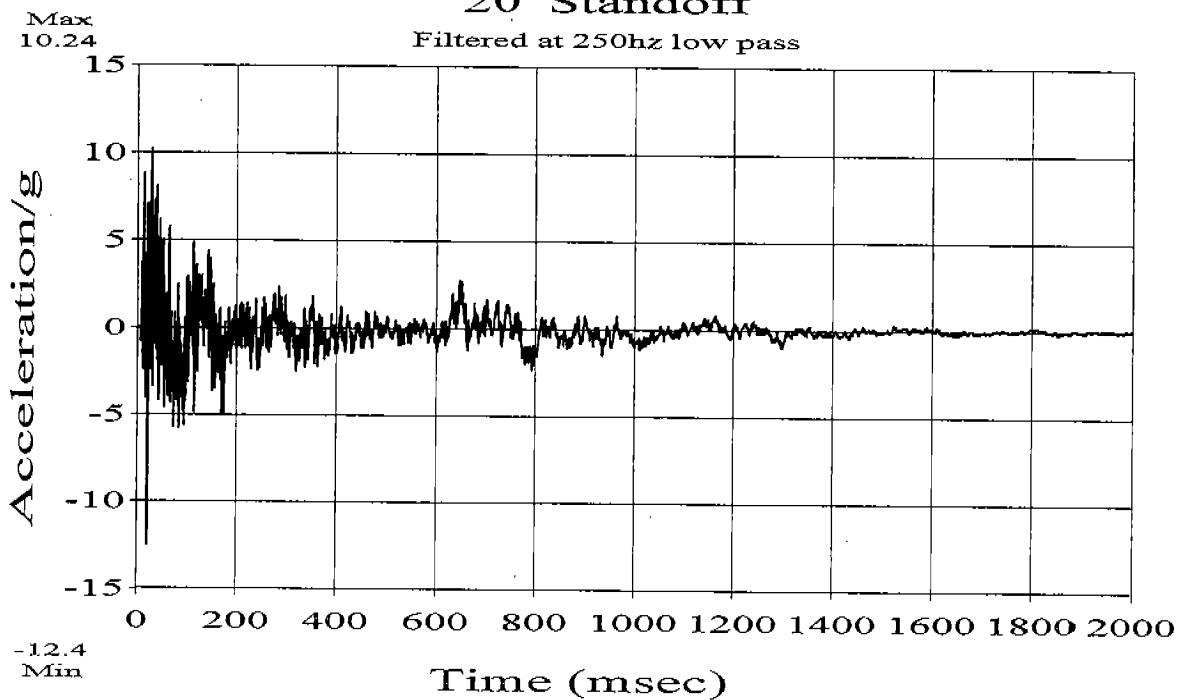
Shot3_Test4_J655

20' Standoff



Shot3_Test4_J655

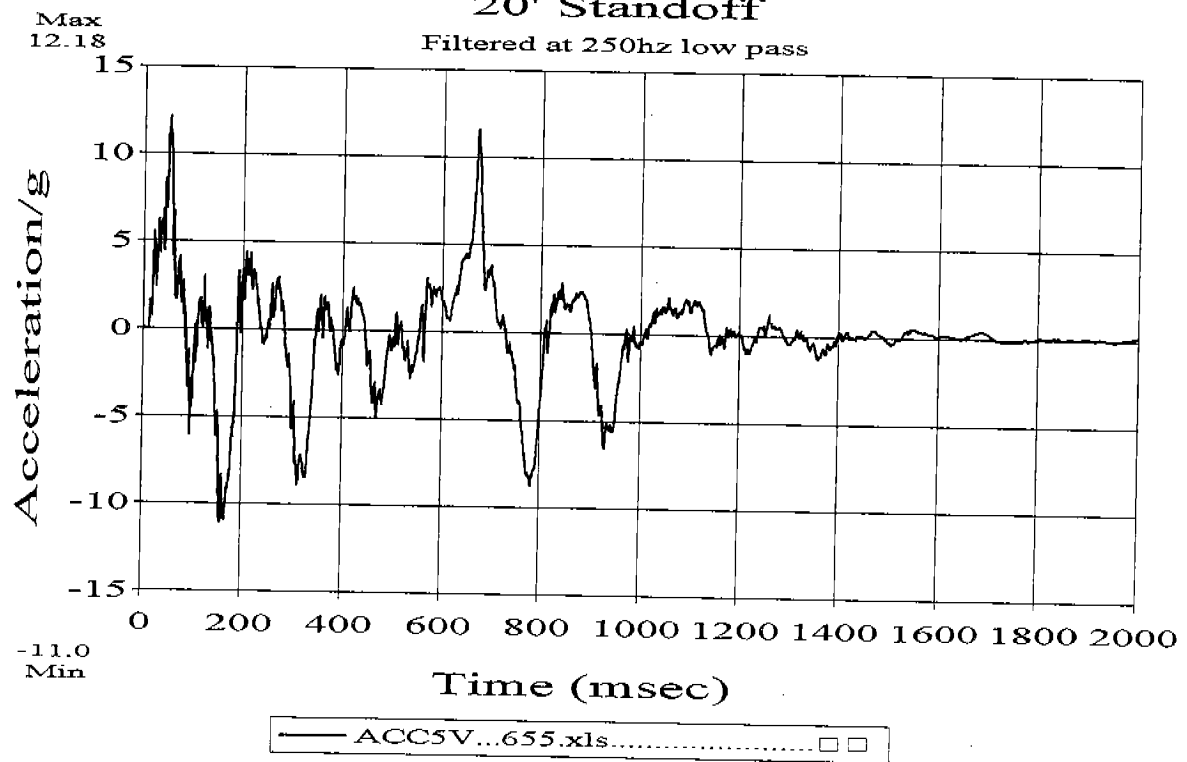
20' Standoff



— ACC2A...655.xls..... □ □

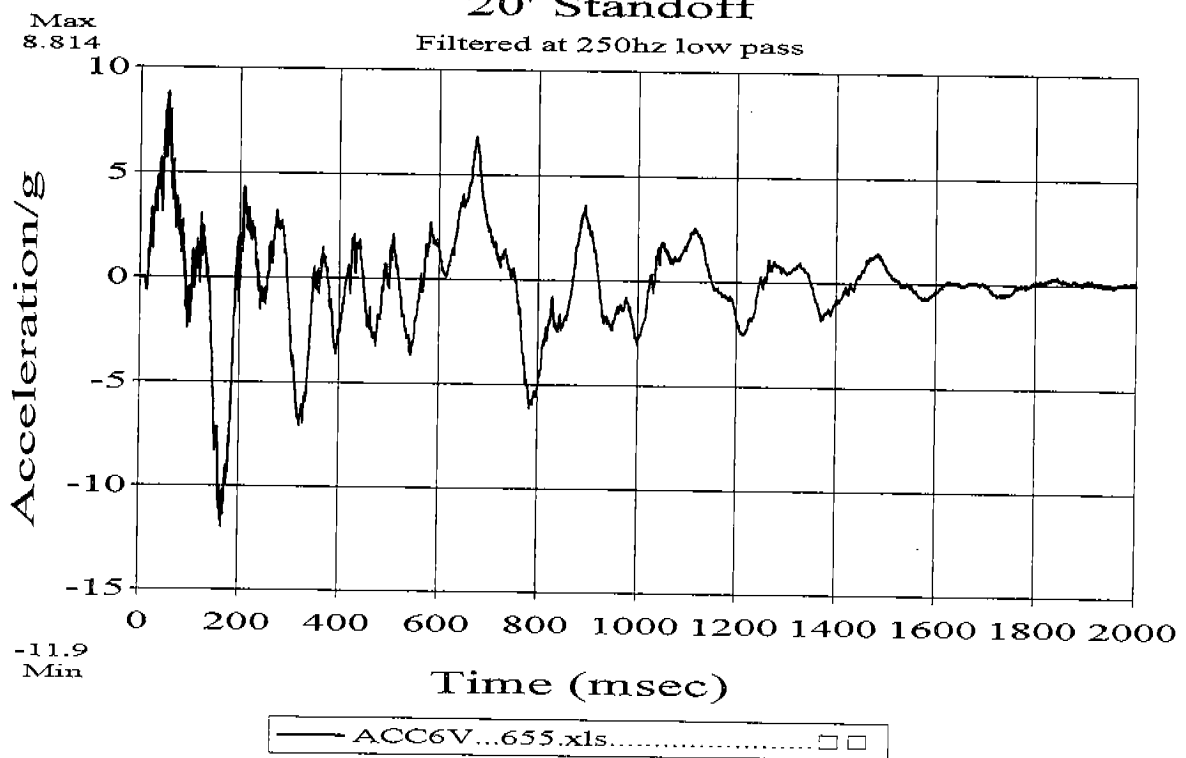
Shot3_Test4_J655

20' Standoff



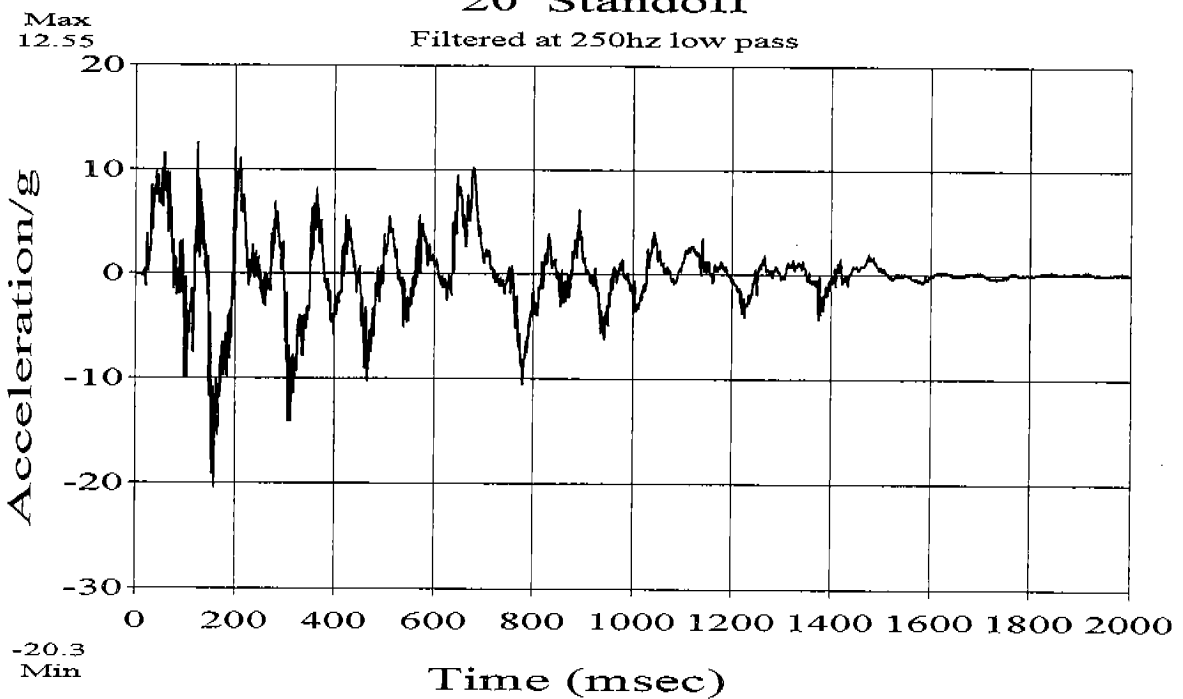
Shot3_Test4_J655

20' Standoff



Shot3_Test4_J655

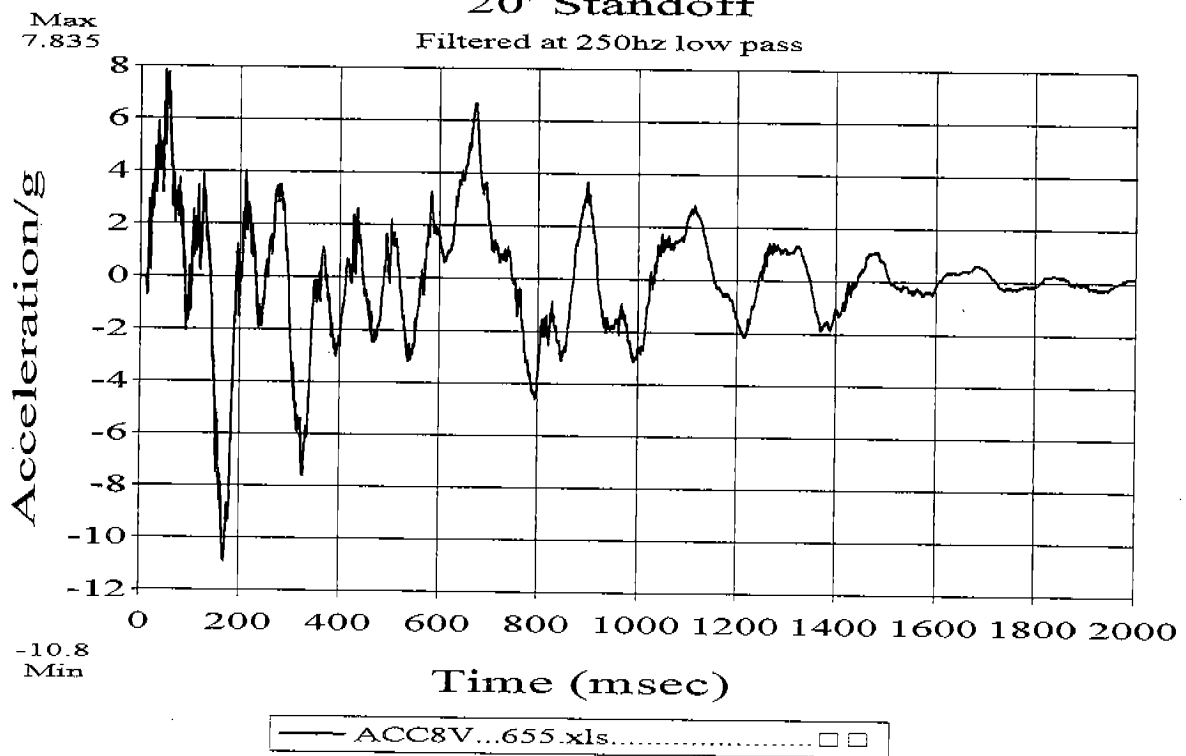
20' Standoff



— ACC7V...655.xls..... ☐ ☐

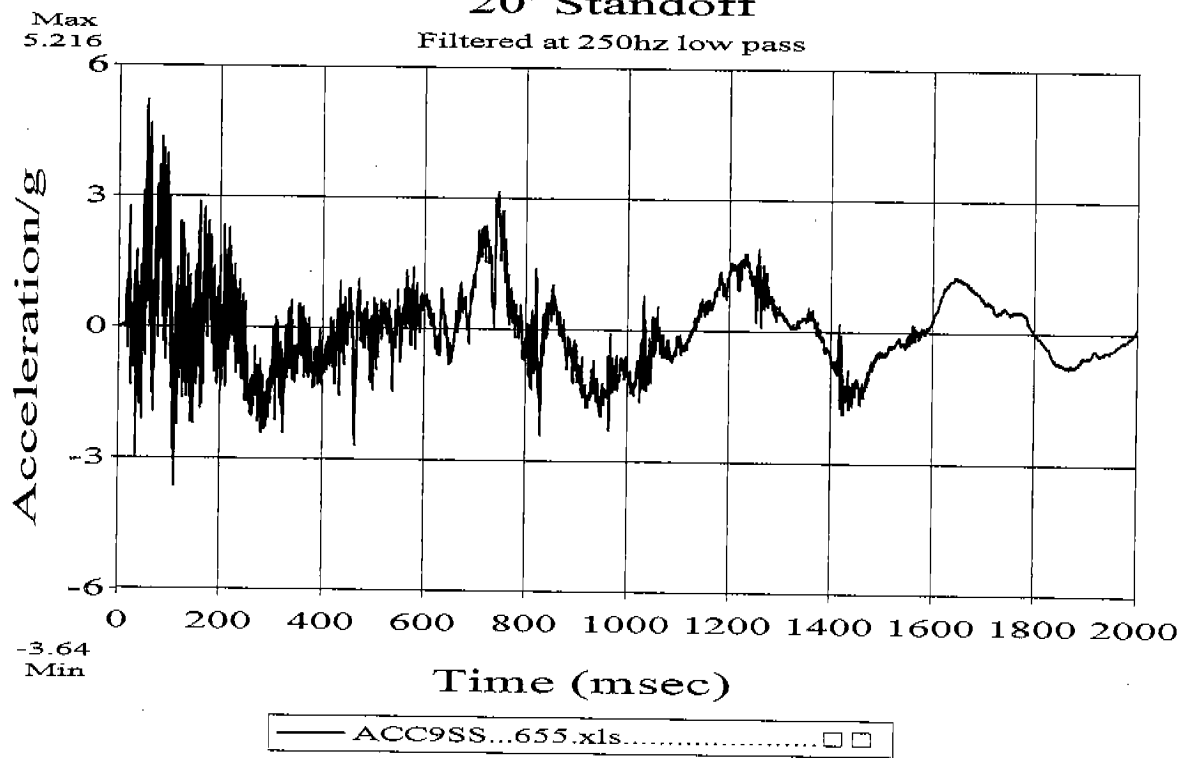
Shot3_Test4_J655

20' Standoff



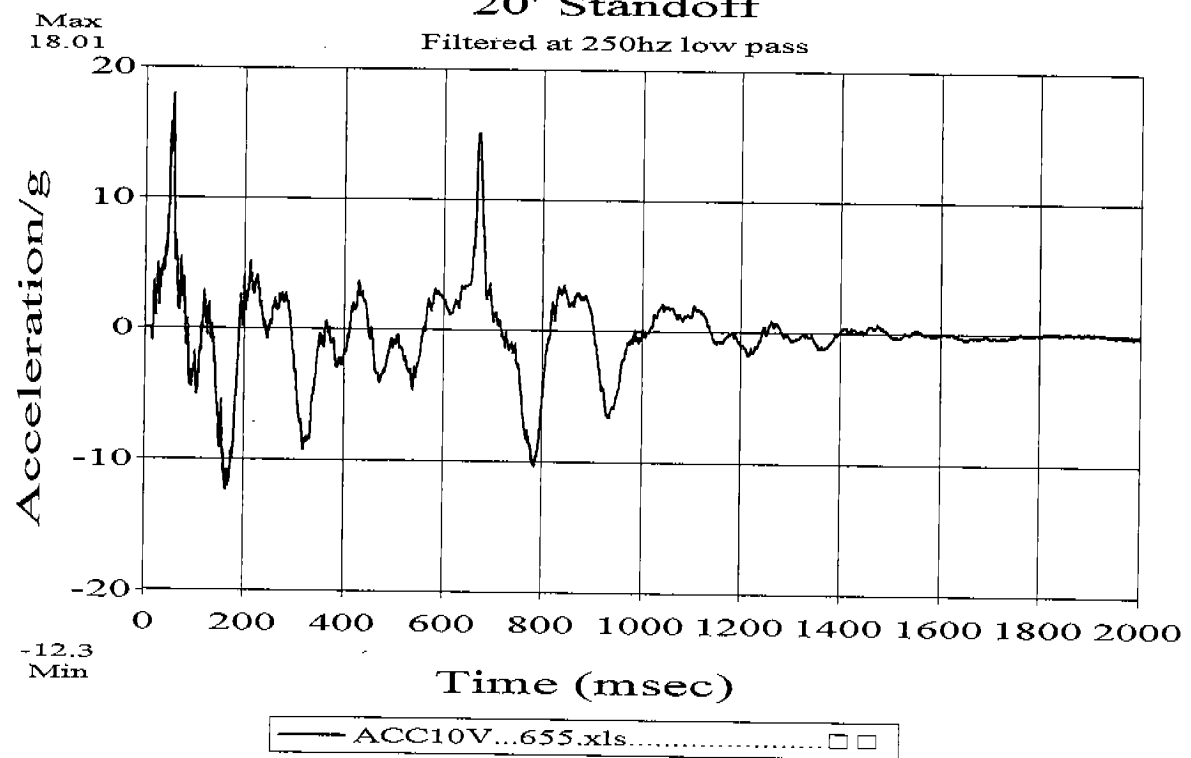
Shot3_Test4_J655

20' Standoff



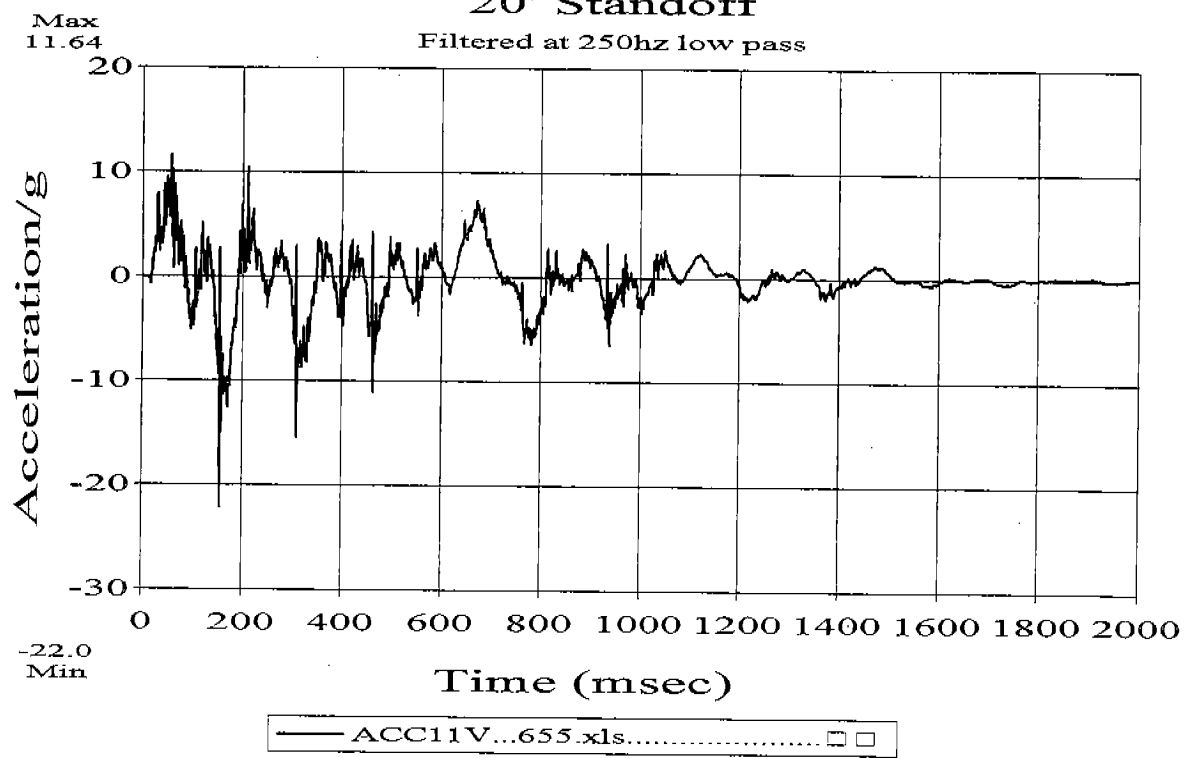
Shot3_Test4_J655

20' Standoff



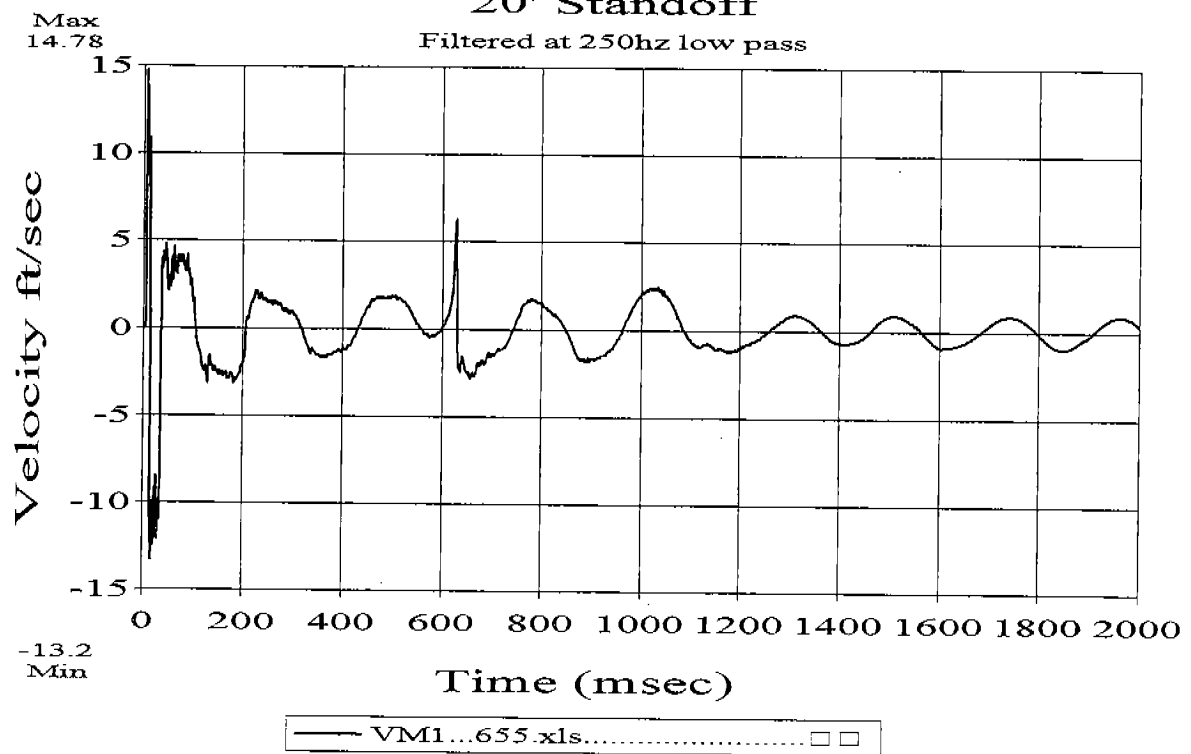
Shot3_Test4_J655

20' Standoff



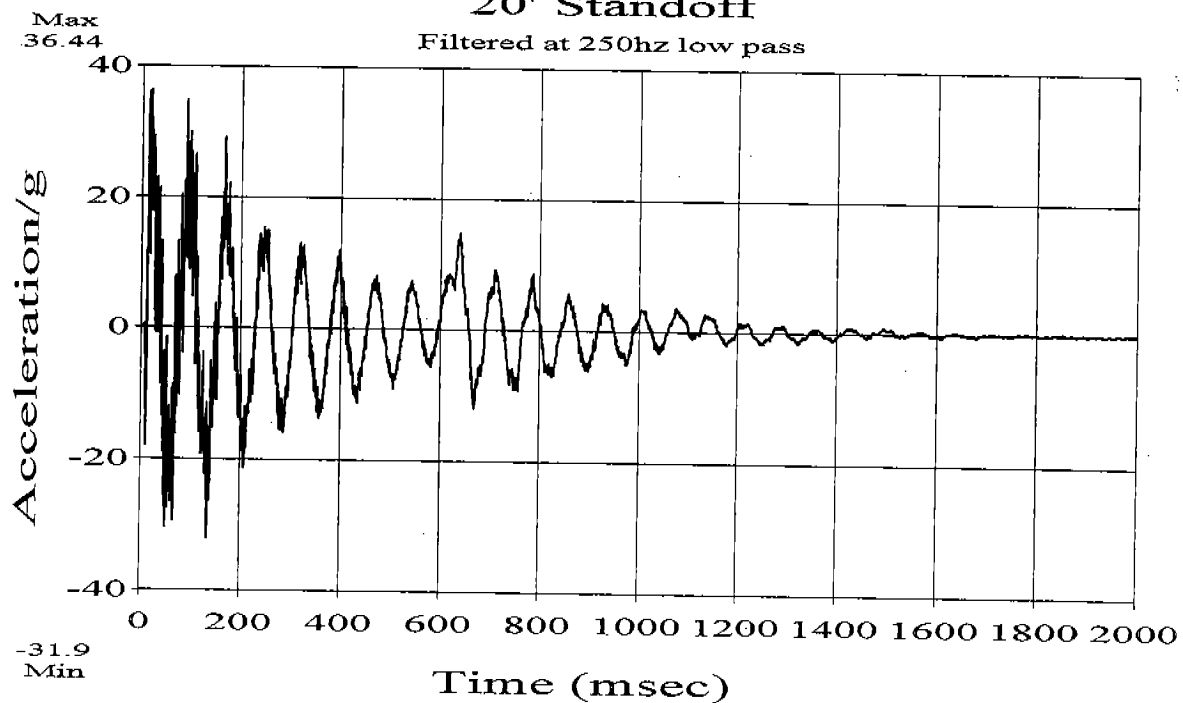
Shot3_Test4_J655

20' Standoff



Shot4_Test4_J655

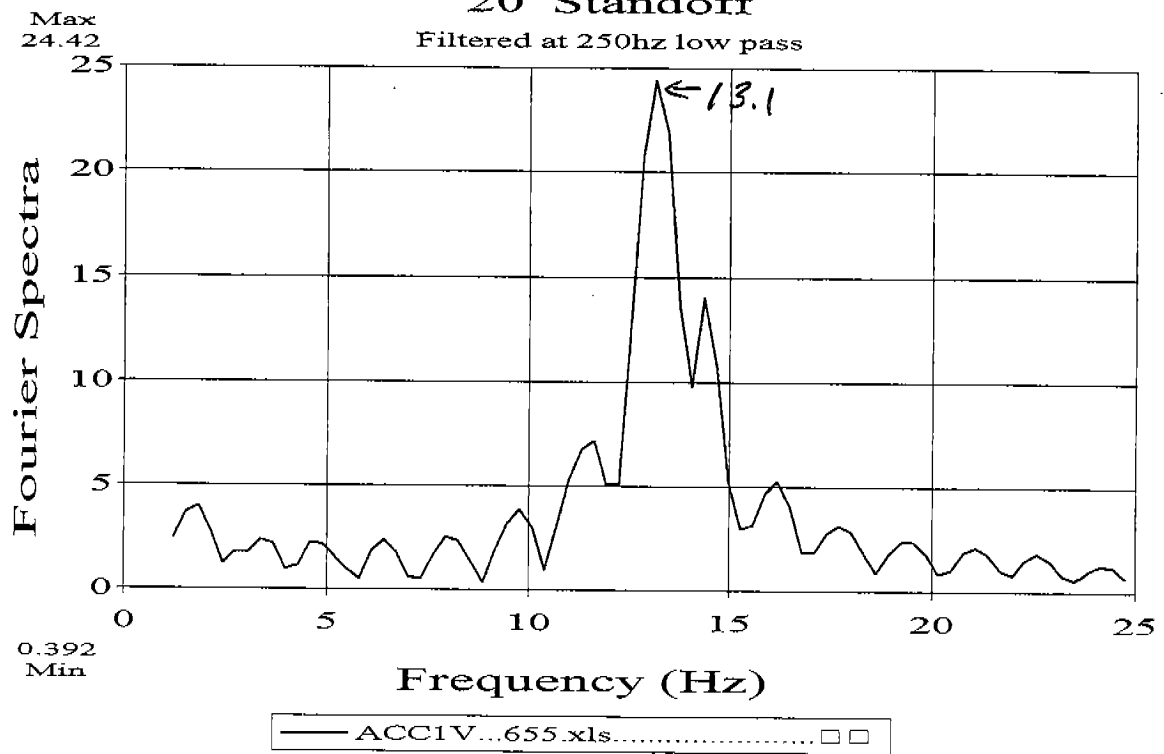
20' Standoff



— ACC1V...655.xls..... □ □

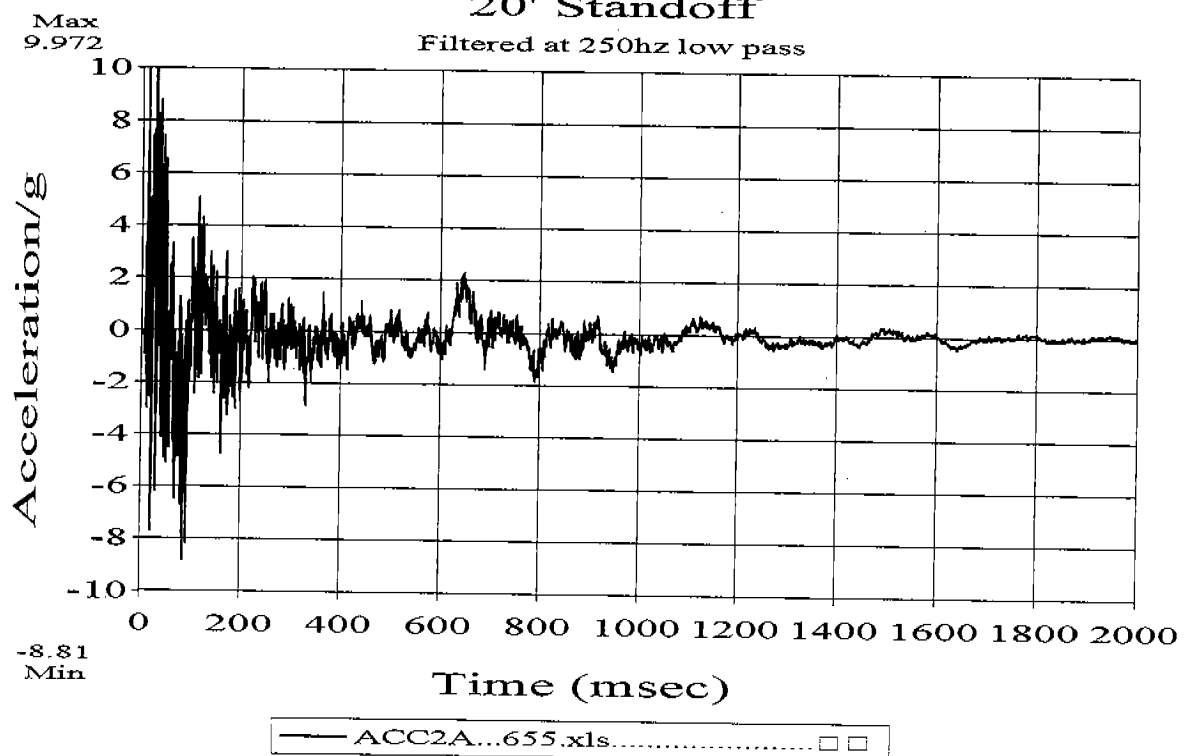
Shot4_Test4_J655

20' Standoff



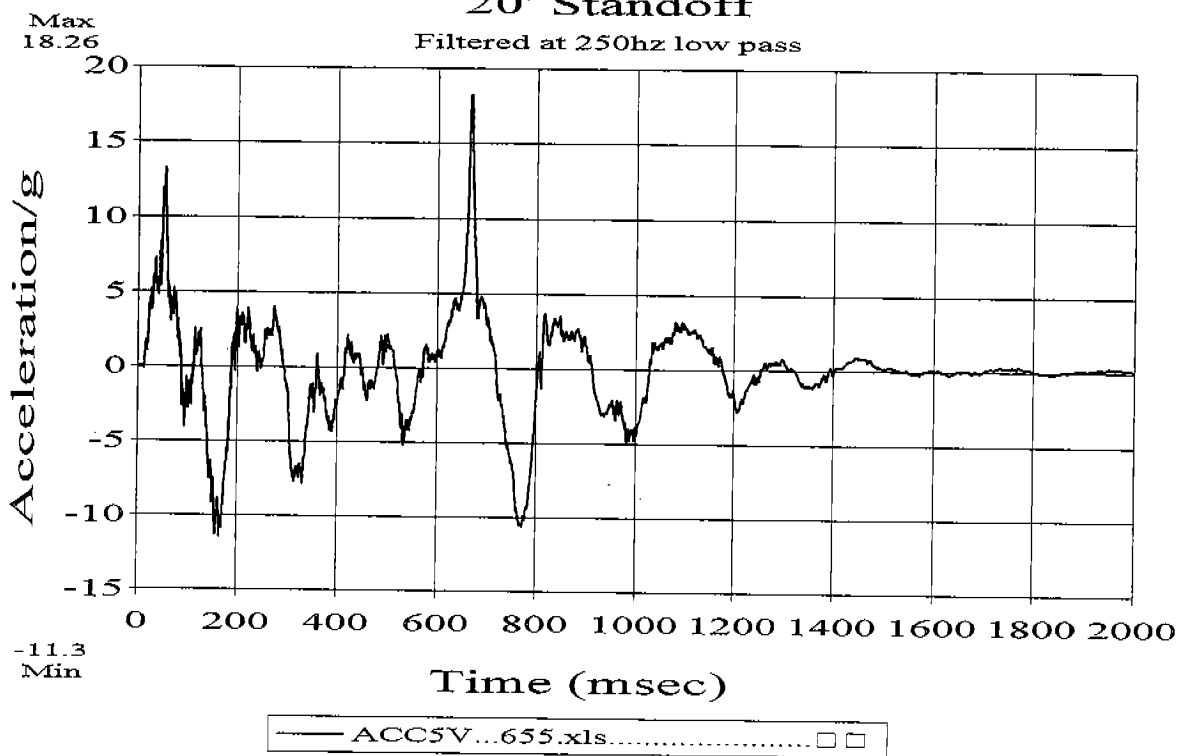
Shot4_Test4_J655

20' Standoff



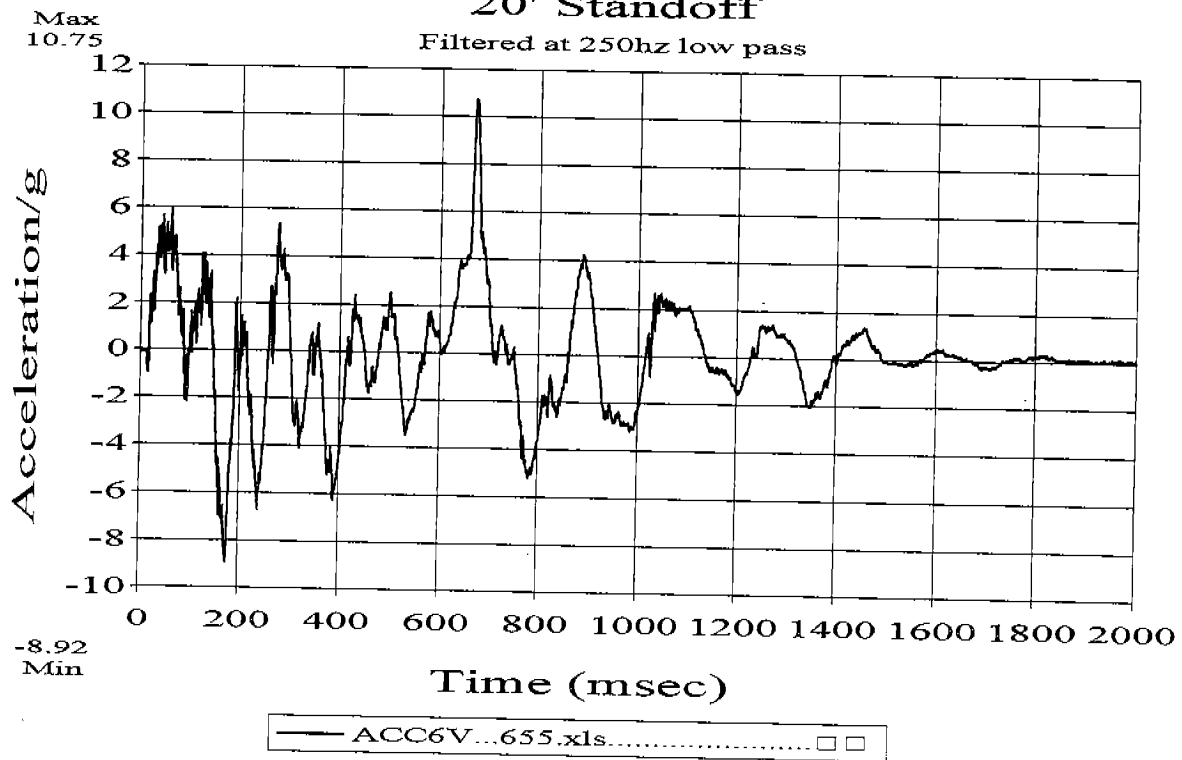
Shot4_Test4_J655

20' Standoff



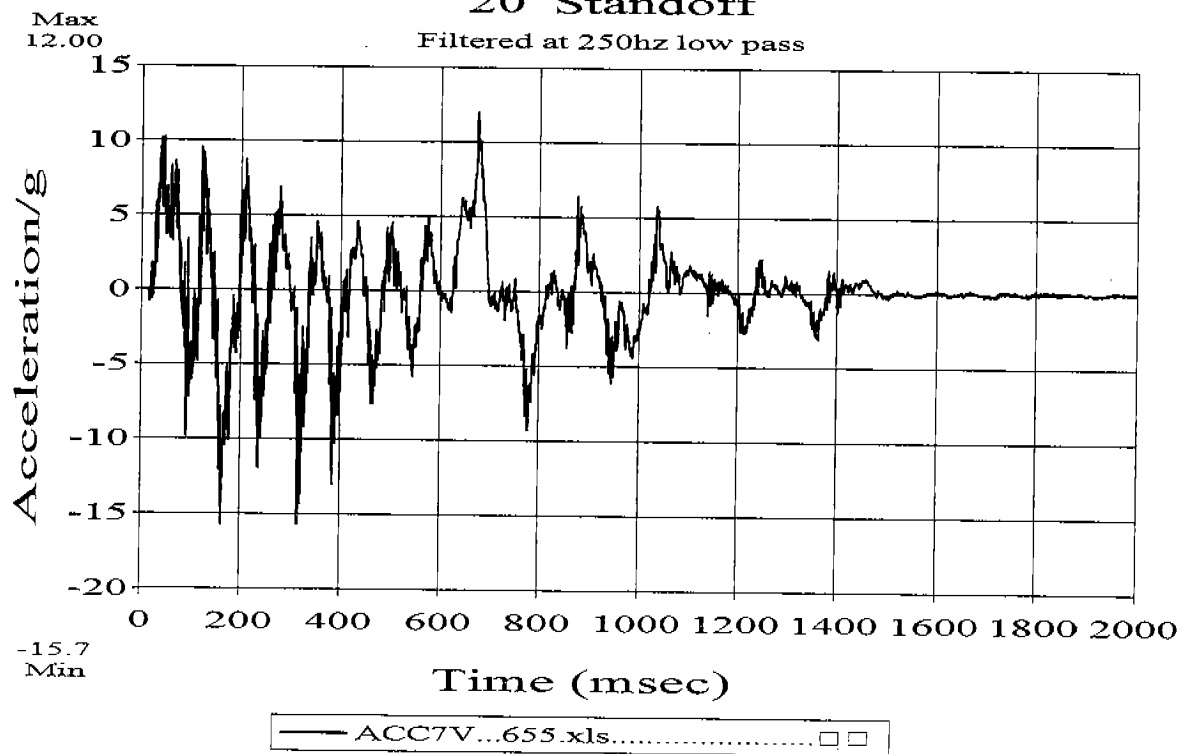
Shot4_Test4_J655

20' Standoff



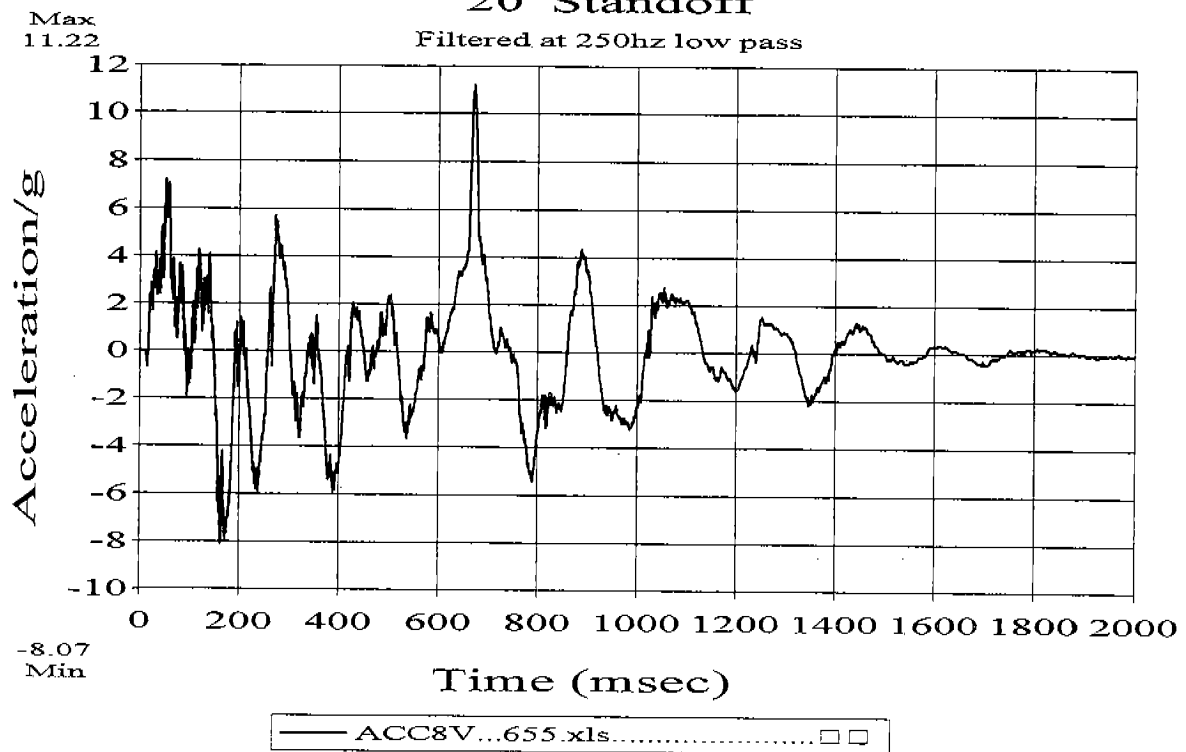
Shot4_Test4_J655

20' Standoff



Shot4_Test4_J655

20' Standoff

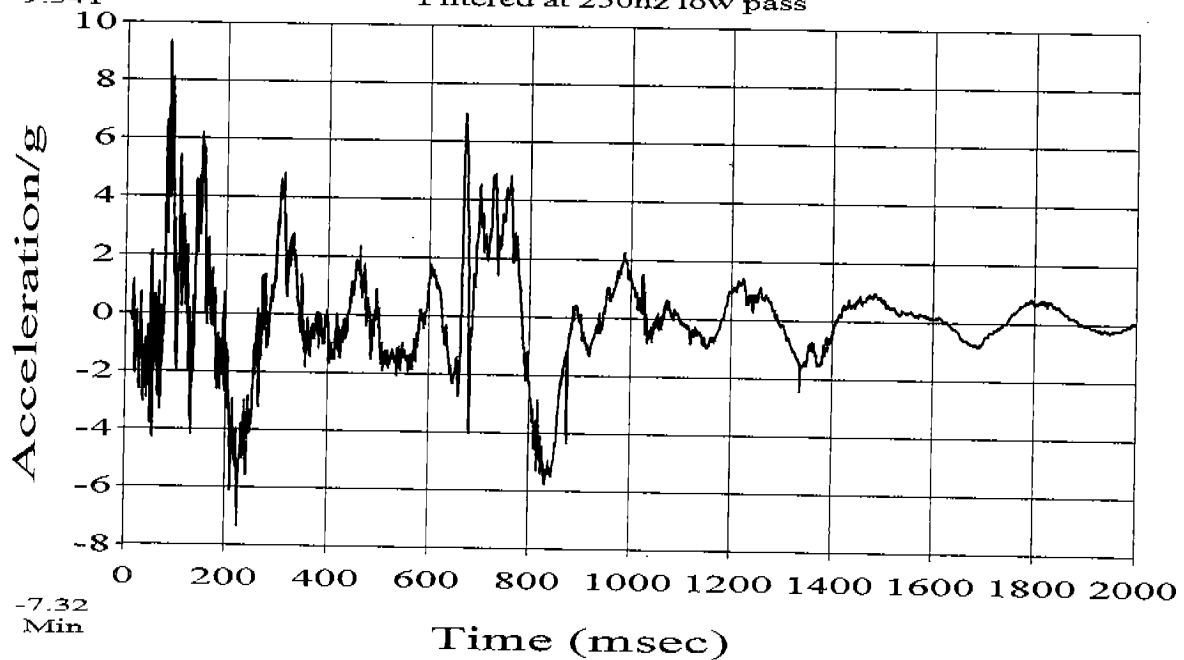


Shot4_Test4_J655

20' Standoff

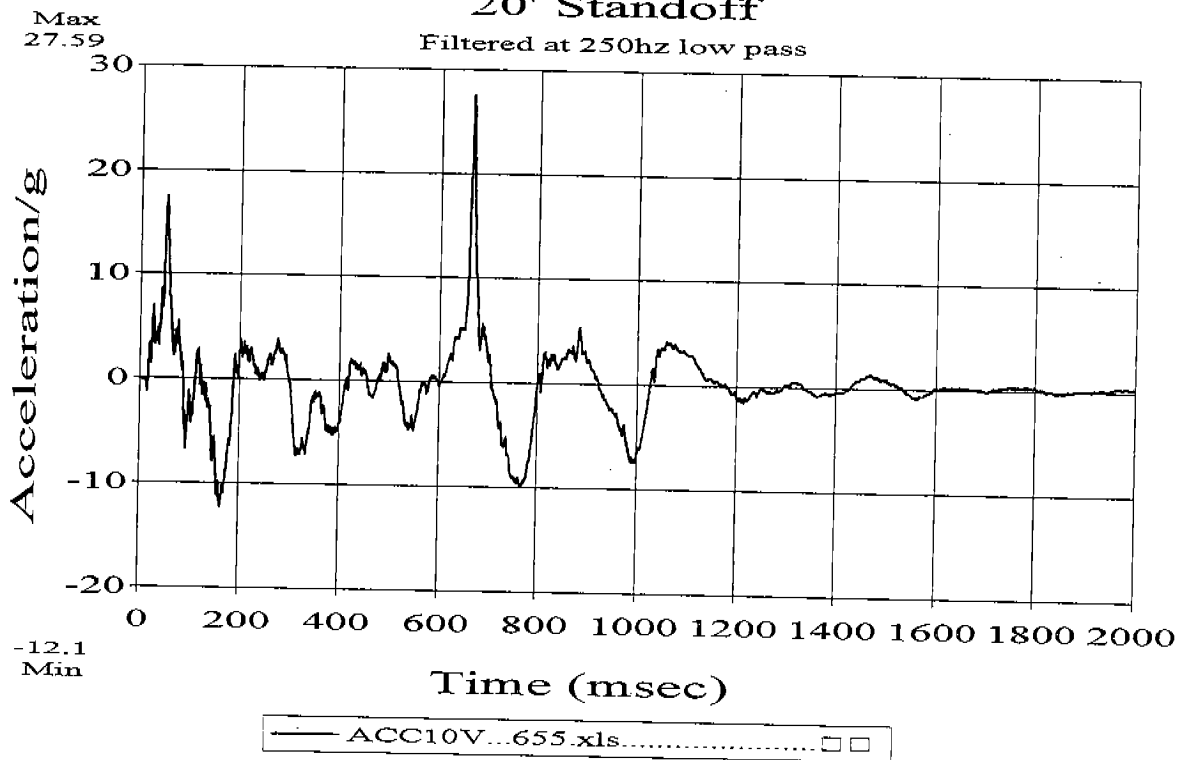
Max
9.341

Filtered at 250hz low pass



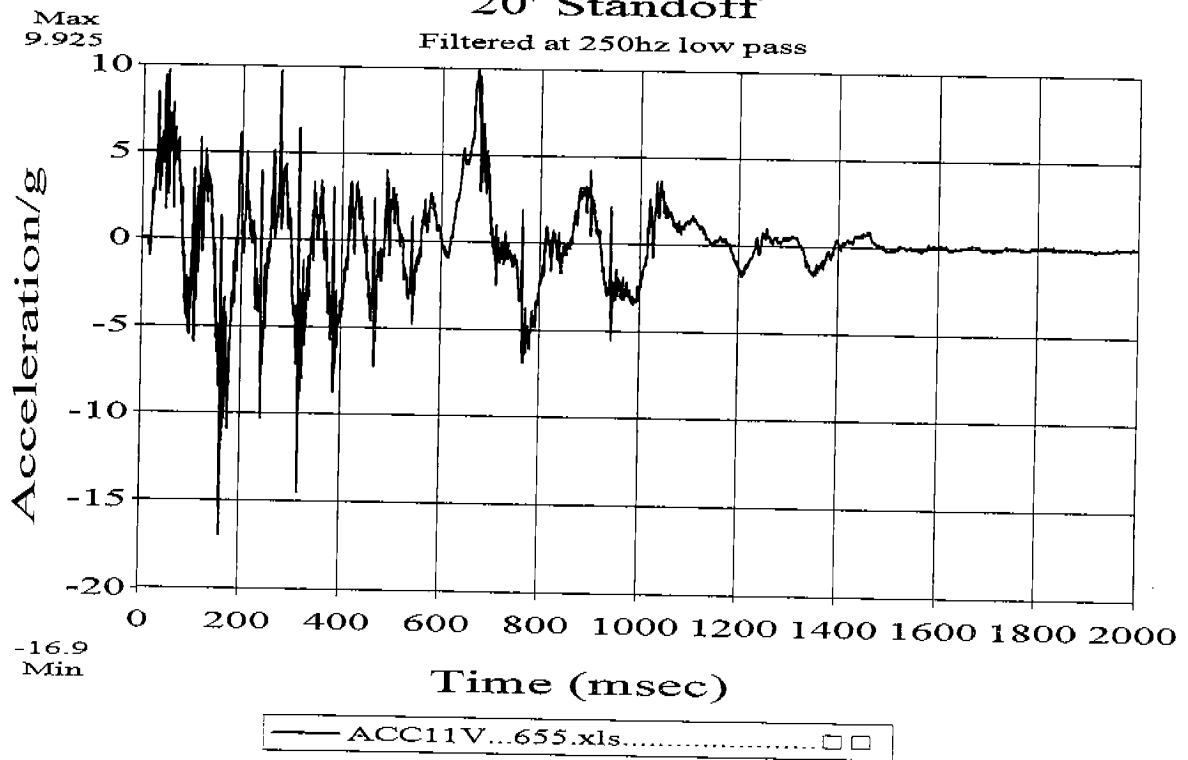
Shot4_Test4_J655

20' Standoff



Shot4_Test4_J655

20' Standoff



Shot4_Test4_J655

20' Standoff

